

## Pitts S-1C, G-BUTO, 8 April 1997

### AAIB Bulletin No: 8/97 Ref: EW/G97/04/03 Category: 1.3

<b>Aircraft Type and Registration:</b>	Pitts S-1C, G-BUTO
<b>No &amp; Type of Engines:</b>	1 Lycoming O-320-A piston engine
<b>Year of Manufacture:</b>	1972
<b>Date &amp; Time (UTC):</b>	8 April 1997 at 1440 hrs
<b>Location:</b>	2nm South East of Barton-Under-Needwood, Staffordshire
<b>Type of Flight:</b>	Private
<b>Persons on Board:</b>	Crew - 1 - Passengers - None
<b>Injuries:</b>	Crew - None - Passengers - N/A
<b>Nature of Damage:</b>	Aircraft destroyed
<b>Commander's Licence:</b>	Airline Transport Pilot's Licence
<b>Commander's Age:</b>	55 years
<b>Commander's Flying Experience:</b>	10,036 hours (of which 95 hours were on type) Last 90 days -107 hours Last 28 days - 38 hours
<b>Information Source:</b>	Aircraft Accident Report Form submitted by the pilot and telephone enquiries by the AAIB

The pilot had been airborne for just over an hour and had completed two aerobatics sequences; this was in preparation for a forthcoming competition. The weather was excellent and the wind was light and from the south at the pilot's operating height of 2,500 feet to 4,000 feet amsl. Prior to his third practice, the pilot changed his location out of consideration to people on the ground; this new location was over ground which was approximately 200 feet amsl.

The aerobatics sequence went as planned until the top of a stall turn. The airspeed was slightly low as the pilot applied full left rudder. He was attempting to complete the manoeuvre on a specific heading and can remember advancing the throttle slightly and applying full forward control column. Almost immediately, the aircraft flicked into an inverted spin. The pilot was surprised at the high rate of descent and cannot recall his precise recovery actions. He can remember that the throttle was fully retarded and that he brought the control column fully back but with the ailerons central. With

full left rudder still applied, the aircraft was not recovering and the pilot thinks that he may then have relaxed his rudder application or even applied some right rudder; he did not check the turn direction from the 'Turn and Slip' indicator. However, he noted the altimeter indicating 2,000 feet amsl. As this was his self-briefed abandonment height, he immediately went for his harness release and bailed out of the aircraft; he was aware of being thrown forcibly out of the cockpit as he released his harness. The parachute descent was uneventful and the pilot landed a few hundred yards from where G-BUTO had crashed.

Several eye witnesses saw the last few moments of flight and the parachute descent. They saw the aircraft do two vertical looping manoeuvres before going into a vertical climb. It was then seen to move erratically from side to side before turning over nose down and spiral towards the ground. The pilot was seen to leave the aircraft and his parachute opened almost immediately.

The pilot had made weight and balance calculations prior to flight and, subsequent to the accident, he rechecked the figures and confirmed that the weight and centre of gravity of the aircraft were within the correct limits. He also stated that the aircraft had been fully serviceable during the flight and acknowledged that it was a mishandled stall turn that caused the loss of control. With the short time between this loss of control and his decision to bail out of G-BUTO, the pilot cannot be certain of his recovery actions; he considers that he may have used some incorrect actions. He has had reasonable experience in aerobatics and has completed training in both upright and inverted spinning but was surprised at how quickly the aircraft entered the spin and the extent of his disorientation.

The pilot acknowledged the value of his habit of wearing a parachute during aerobatics flights and of his pre-planned abandonment height.