Piper PA-23-160, G-ARTD

AAIB Bulletin No: 6/2000 Ref: EW/C99/8/2 Category: 1.3

Aircraft Type and Registration: Piper PA-23-160, G-ARTD

No & Type of Engines: 2 Lycoming O-320-B2B piston engines

Year of Manufacture: 1958

Date & Time (UTC):2 August 1999 at about 1326 hrsLocation:Moel Hebog, North WalesType of Flight:Post engineering test flightPersons on Board:Crew 1 - Passengers - NoneInjuries:Crew - 1 Fatal - Passengers - N/A

Nature of Damage: Aircraft destroyed

Commander's Licence: Basic Commercial Pilots Licence

Commander's Age: 56 years

Commander's Flying Experience: 5,120 hours (of which 67 were on type)

Last 90 days - 136 hours

Last 28 days - 25 hours

Information Source: AAIB Field Investigation

History of flight

The main purpose of the flight was thought to be to an airborne test of the function of the right engine starter motor, which had been refitted after repair. The precise time of departure from Runway 26 at Caernarfon Airport wasnot determined, however, it was the usual practice to call Royal Air Force(RAF) Valley ATC shortly after take off. A transmission was recorded between 1314:50 hrs and 1315:36 hrs, a transcription of which is reproduced, inpart, below:

Aircraft: --- TWINAPACHE JUST AIRBORNE OUT OF CAERNARFON SQUAWKING 7000 IN

THE CLIMB TO THE LLEYNPENINSULA AREA FOR AIRTEST REQUESTING FIS

ATC: --- FISTHE HOLYHEAD 1007 SAY AGAIN HEIGHT

Aircraft: 1007CLIMBING TO 4000 FEET WILL BE OPERATING IN THE BLOCK 4000 TO 2000

FEET

No further radio communication was reported between the aircraft and any ground station.

Radar data, recorded from both St Annesand Clee Hill Radars, was used to determine the aircrafts flight path. The first contact was at 1316:33 hrs, 1.4 nmnorthwest of the airfield; the aircrafts altitude was about1,540 feet amsl. Subsequentcontacts indicated that the aircraft was turning right and climbing. It passed the eastern edge of the airport,tracked southeast and had climbed to about 4,840 feet amsl by 1321:30hrs. Almost immediately it started todescend and when, at 1324:01 hrs, it was 2.3 nm to the south west of Moel Hebogit started a left turn to track northerly towards the mountain. During the turn the

average rate of descentwas about 2,400 fpm and the groundspeed was about 165 to 170 kt. Radar contact was lost at 1325:10 hrs whenthe aircraft was 1.7 nm south of Moel Hebog, at 3,040 feet amsl.

The lack of radar contact, below 1,500 feet amslshortly after take off, and below 3,000 feet amsl just prior to theaccident, was attributed to the intervening high ground between the aircraftand the radar heads.

Witnesses at the eastern end of LlynCwmystradllyn, the lake at the foot of Moel Hebog, saw the aircraft fly overthem, in straight and level flight, into the side of the mountain. The engine sounded normal and there was nochange in the engine noise up to the point of impact. One of these witnesses then went to a house at the other end of the lake, about 0.7 nm away, and telephoned the emergency services. The police logged this call at 1348:15 hrs.

Meteorology

The RAF Valley weather report for 1350 hrswas as follows:

Surface wind 190°/02 kt 2,000 foot wind 190°/10 kt Visibility 15 Km

Weather Nil

Cloud 4 Oktas base 10,000 feet

70ktas base 15,000 feet

QNH/QFE 1011 mb/1009.7 mb

Temp/Dew point 22°C/14°C

Witnesses in the localarea confirmed that there was no low cloud, very little wind and that thevisibility was good.

Pilots flying experience

The pilot started flyingtraining at Peterborough (Sibson) in June 1982. He was, at the time serving as an Air Traffic Control Officer in RAF. He was awarded a PrivatePilots Licence (PPL) in September that year. He started the Assistant Flying Instructor (AFI) Course at Goodwood in April 1987 and was awarded an AFI rating in July 1988. He then started instructing at Goodwoodwhere he remained until May 1989 when he obtained a Group B aircraft ratingand moved to Elstree.

In April 1990 his PPL wasconverted to a Basic Commercial Pilots Licence under transitional arrangements then in force; the conversion involved no tests or examinations.

He was the Chief FlyingInstructor (CFI) at Elstree until December 1995 when he moved to Denham, againas CFI. In August 1997 he was issuedwith an American Airline Transport Pilot Certificate.

He left the RAF in January1998 and moved to Caernarfon to take up the post of Airport Director.

Medical and pathology

Examination of the accident site

At the location where theaircraft had impacted, Moel Hebog sloped upwards generally at about 45° butlocalised rocky crags were nearly vertical. G-ARTD had struck one of these at an altitude of about 1,850 feetamsl, the wreckage having then dropped two metres or so. The only readily identifiable (albeit badlydamaged) part of the airframe was the extreme rear fuselage and the empennagewhich lay in an inverted attitude. The disrupted remains of the cockpit, principally tubular structure, controlcables, piping and wiring were tangled together with the engines and noselanding gear. There was somesheet-metal scatter upwards and to the right of the main site, which generally turned-out to be the remains of both wings and engine cowling panels. Both propellers had detached from the engines and bore characteristic leading edge and tip damage suggesting that roughly cruise power was present at impact.

It was just possible to discern an imprint of the aircraft where it had originally impacted the mountain face. It appeared to be banked about 5° to the right, roughly level in pitch, and had struck the face approximately at right angles. The degree of disruption to the airframe and engines suggested that the aircraft been travelling at a speed in excess of 160 kt.

With some difficulty, thewreckage was recovered from Moel Hebog and brought to the AAIB facility atFarnborough for further examination.

Off-site examination and engineering discussion

As already noted the wreckagewas extremely badly disrupted, rendering conclusive examination of items suchas the flying control runs extremely difficult. Even most of the instrument faces proved difficult to examine oreven identify. Within theselimitations, it can be said that no anomalies were found with the flyingcontrols, structure or engines. The aircraft had originally been fitted with arudimentary autopilot but this had been removed some time ago and only the rollservo, forming part of the control column, remained from the originalinstallation.

The aircraft was fittedwith a Janitrol type heater and this was examined in detail to see whetherthere was any possibility that defects in this could have led to leakage ofcarbon monoxide into the cabin. Evidencewas found that it was not in operation and it is considered unlikely that, given the temperature of the day, the pilot would have found it necessary touse the heater. The low carbon monoxidelevel detected in the pilots body during post mortem examination reinforces the conclusion that this was not a factor in the accident.

Summary

In his radio transmissionshortly after take off, the pilot expressed his intention to climb to4,000 feet amsl and to carry out an airtest. The test of the starter motor would haveinvolved the shutdown and restart of the right engine and the associatedfeathering/unfeathering of its propeller. It would be reasonable to expect the pilot to do this while in levelflight probably at 4,000 feet amsl, the higher level mentioned in theradio transmission. Radar contact wascontinuous between 1316:33 hrs and 1325:10 hrs and showed that the aircraftclimbed through the intended level and shortly afterwards, with no significant period of level flight, entered a continuous descent until contact waslost. This implies that the proposedairtest was not carried out.

The deviation from what could be considered the expected flight regimeoccurred when the aircraft climbed through 4,000 feet amsl. The uninterrupted secondary radar returnsover this period, and for at least a further 4 minutes, confirms the presence of electrical power. It is reasonable to expect the pilot to have made a radio transmission if a technical problemexisted which affected his control of the aircraft. The aircraft was well within radio range of RAF Valley but nosuch transmission was recorded.

The timing suggests thatthere was only a short period between the final radar contact and the pointwhen the eyewitnesses to the impact first saw the aircraft. During this time it had descended from 3,040 feet amsl and was in level flight at about 2,000 feet amsl until it struck the mountainside. They neither saw any flightpath deviation nor heard any change in engine note prior to impact.

Conceiving any possible technicalcauses why an aircraft of this type should fly, ostensibly straight and leveland at cruise power and speed, into a steep mountain face in good visibility isextremely difficult. Disconnection orfreezing of the flying controls would have to occur in all three axes fordeviation from a set flightpath to be impossible. This is considered extremely unlikely as it would have had tooccur during the final seconds of the flight with the aircraft already trimmed straightand level and heading towards the mountain.

The engine noise was constant prior to impact and both enginesappeared to have been producing a similar output, roughly cruise power, atimpact. This implies that the pilotmade no attempt to enter a climb or, if the

extremely unlikely disconnection or freezing of the flying controls in all three axes had occurred, to useasymmetric power to turn the aircraft.

In the absence of evidence of a technical defect, a human factor for the subsequent almost continuous descent into the mountainside in fine weather was considered. The Inquest was held at Caernar fon on 27 April. The Coronersaid that there was insufficient evidence to support any of the findings available to him and therefore recorded an Open verdict.