Piper PA-28RT-201T, G-BNZG, 20 June 1996

AAIB Bulletin No: 11/96 Ref: EW/G96/06/29 Category: 1.3

Aircraft Type and Registration: Piper PA-28RT-201T, G-BNZG

No & Type of Engines: 1 Continental TSIO-360-FB1 piston engine

Year of Manufacture: 1980

Date & Time (UTC): 20 June 1996 at 1044 hrs

Location: Sleap, Shropshire

Type of Flight: Private

Persons on Board: Crew - 2 - Passengers - None

Injuries: Crew - None - Passengers - N/A

Nature of Damage:

Damage to cowling, nose landing gears doors and

propeller

Commander's Licence: Commercial Pilot's Licence

Commander's Age: 28 years

Commander's Flying Experience: 1,950 hours (of which 120 were on type)

Last 90 days - 150 hours

Last 28 days - 66 hours

Information Source: Aircraft Accident Report Form submitted by the pilot

The aircraft was engaged on a circuit training detail with the student handling. The instructor reported that, on the firstapproach to land, the student was controlling the airspeed welland that he was following a 3° glidepath. He further added that all three landing gear locked-down green lights were illuminated. Touchdown on the mainwheels on Runway 05 was gentle at approximately 60 kt and the nose was lowered. However after rolling some 30 metres on all 3 wheels, the crew realised that the nose landinggear was slowly starting to collapse and the propeller contacted the ground.

The instructor took control and shut down the aircraft systems it rolled to a stop with the nose on the runway. There wasno fire and the aircraft was evacuated without difficulty. Uponexamination, it was found that the downlock hook (see diagram)had broken and that the roller upon which it engages was badlybent. It was evident that the distorted roller had contacted the exterior surface of the hook and caused its detachment, therebeing no evidence of pre-existing cracks in the hook

fitting whichwas recovered from the runway. It appears possible that the rollermay have been bent for some time, but still able to enter thehook. On this occasion it is postulated that it might have rotated and been presented to the hook in an adverse orientation, causing the damage. The maintenance company could not explain how thepin distortion had occurred, but recalled another occurrence to a different aircraft where a stone, or a piece of concrete, hadlodged in the mechanism and led to a similar effect.

Since the detached part of the hook contained the striker platefor the microswitch controlling the green downlock light, it appears that the hook must have been in its correct orientation during the final approach, although possibly close to failure. A piloton the previous flight of this aircraft reported that he did notice that the nose gear green light momentarily went out during taxying, but as it had re-appeared he did not think it worthy of comment.