

No: 8/88

ERRATA

Bulletin 7/88

Page 3 - Boeing 747-136, G-AWNB

Last paragraph should read "Boeing developed an ultrasonic technique for crack detection of the beam end fitting (NDT Manual, D6-7170, Part 4 Chapter 57-40-03 revision 15 Feb 88) and developed a rework procedure and limits which appeared in Service Letter 747-SL-57-50-A (11 December and subsequent). A Service Bulletin (747-57-2244) issued on 31 March urged operators to carry out the described ultrasonic crack detection procedure upon the accumulation of 30,000 flight hours or 8 years service (whichever earlier) or within 18 months after receipt of the bulletin, whichever later. The bulletin also described the reworking procedure to be carried out if cracking were detected and which also could be implemented to terminate a requirement for 18 monthly repeat inspections. From airframe 696 onward bushings and bolts in the beam end assembly will be assembled with wet sealant and fillet sealed where appropriate to improve corrosion protection".