

# Agusta Bell 206B, G-PEAK, 2 September 1997

## AAIB Bulletin No: 4/98 Ref: EW/G97/09/01 Category: 2.3

<b>Aircraft Type and Registration:</b>	Agusta Bell 206B, G-PEAK
<b>No &amp; Type of Engines:</b>	1 Allison 250-C20 turbine engine
<b>Year of Manufacture:</b>	1970
<b>Date &amp; Time (UTC):</b>	2 September 1997
<b>Location:</b>	Maintenance
<b>Type of Flight:</b>	N/A
<b>Persons on Board:</b>	Crew - Nil - Passengers - Nil
<b>Injuries:</b>	Crew - N/A - Passengers - N/A
<b>Nature of Damage:</b>	Cracks in main rotor pitch change link
<b>Commander's Licence:</b>	N/A
<b>Commander's Age:</b>	N/A
<b>Commander's Flying Experience:</b>	N/A
<b>Information Source:</b>	Mandatory Occurrence Report and further inquiries by AAIB

During routine maintenance on a 100 hour check, inspection revealed a suspected crack in one of the main rotor pitch change links at the lower swaged end. Localised paint removal confirmed two cracks approximately 7 mm in length running along the rod from the lower end. The rod was retrieved by AAIB for examination to identify the failure mechanism and to consider the failure's potential to explain an accident to another Agusta Bell 206B.

Detailed examination of the rod confirmed the presence of three small, longitudinal, through-thickness cracks in the rod wall adjacent to the threaded end fitting. (Figures 1 and 2). The cracks measured approximately 2, 7 and 8.5 mm in length and were located at 70° intervals in one half of the rod circumference. The orientation and appearance of the cracks and the presence of corrosion deposits, both within the cracks and in the gap between the end fitting and the rod wall, suggests that a corrosion mechanism was responsible for the cracking observed.

Cracking of this sort was the subject of a Bell Helicopter Alert Service Bulletin and subsequent Airworthiness Directive in 1973 and is now monitored through routine maintenance inspections, the process by which these cracks were identified.

Examination of the main rotor pitch change links from the other Agusta Bell 206B under investigation by AAIB has precluded this cracking mechanism from contributing to that accident.