

# Cessna F172H, G-AZZP

## AAIB Bulletin No: 9/97 Ref: EW/G97/06/04 Category: 1.3

<b>Aircraft Type and Registration:</b>	Cessna F172H, G-AZZP
<b>No &amp; Type of Engines:</b>	1 Rolls Royce Continental O-300-D piston engine
<b>Year of Manufacture:</b>	1970
<b>Date &amp; Time (UTC):</b>	8 June 1997 at 1230 hrs
<b>Location:</b>	Near Headcorn Airfield, Kent
<b>Type of Flight:</b>	Private
<b>Persons on Board:</b>	Crew - 1 - Passengers - 3
<b>Injuries:</b>	Crew - None - Passengers - 2 (Minor)
<b>Nature of Damage:</b>	Wings, nose gear and fuselage (beyond repair)
<b>Commander's Licence:</b>	Airline Transport Pilot's Licence with Flying Instructor Rating
<b>Commander's Age:</b>	34 years
<b>Commander's Flying Experience:</b>	5,200 hours (of which 215 were on type) Last 90 days - 180 hours Last 28 days - 60 hours
<b>Information Source:</b>	Aircraft Accident Report Form submitted by the pilot and telephone enquiries

The Cessna 172 has a very simple fuel system. One tank in each wing is connected to a selector valve which has four positions: BOTH, LEFT, RIGHT and OFF. Fuel is gravity fed from both wing tanks to the carburettor via the fuel strainer and there is no electric fuel pump. An engine primer which draws its fuel from the strainer, may be used by the pilot to obtain a rich mixture for starting a cold engine. A fuel vent pipe which vents both tanks is connected to a vent on the left wing. When full this aircraft's tanks held 191 litres of usable fuel; the block fuel consumption rate was approximately 30 litres per hour.

The pilot arranged to meet with friends at Headcorn airfield and to take them on a short flight around the local area. After briefing and seating his passengers he carried out a pre-flight check which included a visual inspection of the contents of both fuel tanks. Each tank was at least one quarter full and the levels were equal giving a total of at least 45 litres. The pilot was content that this was ample for the 30 minute flight and he decided not to refuel the aircraft because he was

concerned about take-off weight with four persons on board. The MTOW for this type is 2,300 lb; the actual take-off weight was 2,170 lb.

The aircraft took off at 1305 hrs in CAVOK conditions with a light south westerly wind. In the Folkestone area the pilot checked the fuel gauges and noticed nothing amiss or any imbalance. Shortly afterwards the aircraft returned to Headcorn circuit and positioned onto final approach for Runway 29. At a height of about 500 feet on final approach the engine faltered and stopped. The pilot transmitted a MAYDAY message stating that the aircraft had run out of fuel. With no time for detailed failure analysis, he determined that he could not reach the airfield and so he turned downwind and selected a field for a forced landing. The field was quite small and although the aircraft touched down near the start, it overran the available length and came to a sudden stop in the hedge at the far end. The pilot turned off the fuel selector, shut down the electrics, and turned off the ignition before he and his passengers vacated through the right hand exit door.

The airport emergency vehicle arrived at the scene within two or three minutes; the local authority fire and ambulance services arrived soon afterwards. The two front seat occupants were unhurt but the rear seat passengers were taken to hospital with possible injuries. Both were discharged the same day suffering from bruises inflicted by their lap straps.

A witness on the airfield had seen the aircraft in difficulty and immediately ran to the airfield emergency vehicle in which he proceeded to the crash scene with a colleague. On arrival he confirmed that the aircraft's fuel and electrical systems had been made safe and he remained with the aircraft until it was recovered to a hangar on the airfield. After the accident the aircraft's wings were inclined about 30° left wing low and fuel was dripping steadily from the vent pipe on the left wing. The witness estimated that fuel remained dripping for about 25 minutes until the fire service managed to stem the flow. Later, whilst the wings were being removed from the fuselage, the witness collected two gallons of fuel in a bucket which came from within the left wing structure. After the aircraft had been recovered to the owner's maintenance facility, a further five gallons of fuel was drained from the left wing tank.

The pilot stated that he always flew with the fuel selector at BOTH unless there was a compelling need to select a particular tank to restore an imbalance; in his experience this was seldom if ever necessary. He checked that the selector was at BOTH before take-off and had not touched it during flight. He had not used the engine primer either because the aircraft had landed a few minutes before he took off and the engine was still warm.

Because of the inclined wings, the design of the fuel system, and internal damage to the wing structure, it was not possible to determine from which tank the fuel had leaked. Fuel from the right tank could have migrated to the left via the vent pipe.