No: 8/91

Ref: EW/G91/05/11

Category: 2c

Aircraft Type and Registration:

Robinson R22 Beta, G-OVNR

No & Type of Engines:

1 Lycoming O-320-B2C piston engine

Year of Manufacture:

1990

Date & Time (UTC):

26 May 1991 at 1530 hrs

Location:

Harrietsham, Kent

Type of Flight:

Private

Persons on Board:

Crew - 1

Passengers - 1

Injuries:

Crew - Minor

Passengers - Minor

Nature of Damage:

Substantial to landing gear cabin structure and tail boom

Commander's Licence:

Private Pilot's Licences (A and H)

with IMC and Night ratings

Commander's Age:

45 years

Commander's Flying Experience: 87 hours rotary wing (all on type)

181 hours fixed wing

Information Source:

Aircraft Accident Report Form submitted by the pilot and

further investigation by the AAIB

The helicopter was fitted with a rotor speed governor which, when engaged, controlled the rotor speed at $102\% \pm 2\%$. The purpose of the flight was to view a house belonging to the passenger and following an uneventful transit, one circuit of the house was completed at about 900 feet agl. When turning into wind (060°/10 kts) for a further orbit at an indicated airspeed of 5-10 kts, the aircraft suddenly yawed to the right. The pilot applied full left pedal but the helicopter continued to rotate rapidly to the right and started to descend. The pilot applied forward cyclic stick in an attempt to gain airspeed but the aircraft continued its rapid rotation. At a height estimated by the pilot as 50 to 75 feet agl the rotation started to slow down. The pilot flared the aircraft and raised the collective lever before the aircraft struck the ground in a level attitude with considerable vertical velocity and no forward speed. The rotor speed governor had maintained the rotor speed at 102% throughout the incident. The aircraft remained upright after the impact and the engine continued to run. Despite minor injury, the pilot was able to shut down the engine after which both he and his passenger left the aircraft without difficulty.

Inspection of the aircraft by an approved maintenance organisation after the accident indicated that there had been no pre-impact failure that could have caused or contributed to the accident. Engineering evidence also confirmed the pilot's statement that the governor had maintained rotor speed up to the point of the impact.

fear of Manufacture:

Date & Fime (UTC):

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Date & Fime (UTC):

Private

Private

Crew - I

Passengers - I

Injuries:

Crew - Minor

Passengers - Minor

Substantial to landing gear cabin structure and tail boom

with IMC and Night ratings

Commander's Age:

A5 years

Commander's Flying Experience:

B7 hours rotary wing (all on type)

131 hours fixed wing

A6 intait Accident Report Form submitted by the fidot and

Information Source:

A6 hierart Accident Report Form submitted by the fidot and

Information by the AMB

The helic open was fitted with a rotor speed governor which, when engaged, controlled the rotor speed at 10.2% ± 2%. The purpose of the flight was to view a house belonging to the passenger and following an uneventful transit, one circuit of the house was completed at about 900 feet agl. When turning one wind (060%10 kts) for a further orbit at an indicated airspeed of 5-10 kts, the aircraft suddenty yaved to the right. The pilot applied full left pedal but the helicopter continued to rotate rapid by to the right and started to descend. The pilot applied forward cyclic stick in an attempt to gain airspeed but the aircraft continued as rapid rotation. At a height estimated by the pilot as 50 to 75 feet agit the rotation started to slow down. The pilot flared the aircraft and raised the collective lever before the aircraft strack the ground in a level arithde with considerable vertical velocity and no forward speed powersor had maintained the rotor speed at 102% throughout the incident. The aircraft remained upright after the impact and the ongue continued to run. Despite minor injury, the pilot was able to shut down the angine after which both he and his passenger ich the aircraft without