

**ACCIDENT**

<b>Aircraft Type and Registration:</b>	Pegasus XL-R, G-MTKG	
<b>No &amp; Type of Engines:</b>	1 Rotax 447 piston engine	
<b>Year of Manufacture:</b>	1987	
<b>Date &amp; Time (UTC):</b>	9 February 2009 at 1325 hrs	
<b>Location:</b>	Caernarfon Aerodrome, Wales	
<b>Type of Flight:</b>	Private	
<b>Persons on Board:</b>	Crew - 1	Passengers - None
<b>Injuries:</b>	Crew - None	Passengers - N/A
<b>Nature of Damage:</b>	Damage to nosewheel forks, wing keel and propeller	
<b>Commander's Licence:</b>	National Private Pilot's Licence	
<b>Commander's Age:</b>	63 years	
<b>Commander's Flying Experience:</b>	89 hours (of which 34 were on type) Last 90 days - 12 hours Last 28 days - 9 hours	
<b>Information Source:</b>	Aircraft Accident Report Form submitted by the pilot	

**Synopsis**

The flexwing microlight aircraft landed heavily, bounced and rolled onto its side. The pilot was uninjured.

**History of the flight**

The pilot made an overhead join for a landing on Runway 02, which is 1,080 m long. The surface wind was from 040° at 10 kt and the aircraft was turned onto final at a height of 800 ft approximately ¼ mile from the threshold. The pilot initially commenced a glide approach, but realised that he would not reach the runway and therefore applied some power. During the later stages of the approach he realised that he was being blown off the runway centreline and so commenced a go-around. However, as the aircraft regained the centreline he decided to continue with the

landing and reduced the power to continue with the glide approach. The aircraft subsequently landed heavily, bounced and rolled onto its side. The pilot was uninjured and was able to use the aircraft radio to inform the Tower of his condition.

The pilot believes that the accident occurred because he was not stable on the approach and that he should have continued with the go-around. He also informed the AAIB that he intends to take further instruction on crosswind landing techniques.

**Comment**

The BMAA advised the AAIB that the Pegasus XL-R has limited crosswind capabilities and poor energy

retention. However, the crosswind component at the time of the accident was low, at around 3 kt, and therefore should not have been a problem during the approach and landing. It is likely that in the later stages

of the glide approach the airspeed decreased and a high sink rate developed, which resulted in the aircraft landing heavily.