HAWK T MK 1/1A

FLIGHT REFERENCE CARDS

INCORPORATING

AIRCREW LANDAWAY FLIGHT SERVICING SCHEDULE

Amendment Information

Note: This is an electronic version of this publication, and not a direct duplicate of the hard copy version. The hard copy is the master copy and the only one to be used in flight and for flight planning. This electronic publication represents the extant version of the publication and is for reference only. Only change bars applicable to the current version will be shown.

Issue

Issue 6

AL

3 (Oct 17)

ANA

ANA 10

ANA 11

HAWK TMK 1/1A



FLIGHT REFERENCE CARDS

INCORPORATING

AIRCREW LANDAWAY FLIGHT SERVICING SCHEDULE

AIRCRAFT SAFE FOR PARKING

The aircraft is safe for parking when the following are correctly inserted:

Both ejection seat firing handle safety pins Both canopy MDC firing handle safety pins Both canopy MDC firing unit safety pins

These FRCs are only valid when used in support of a current MOD Release to Service

Prepared by Handling Squadron

LIST OF CARDS ISSUE 6 AT AL3

Card	AL	Card	AL	Card	AL	Card	AL
N-1	AL 3	N-31	AL 3	E-11	AL 3		
N-3	AL 3	N-33	AL 3	E-13	Issue 6		
N-5	AL 3	N-35	AL 3	E-15	AL 3		
N-7	AL 3	N-37	Issue 6	E-17	AL 3		
N-9	Issue 6	S-1	AL 3	E-19	AL 3		
N-11	AL 3	S-3	AL 1	E-21	Issue 6		
N-13	AL 3	S-5	AL 3	E-23	Issue 6		
N-15	AL 3	S-7	AL 1	E-25	AL 2		
N-17	Issue 6	S-9	AL 3				
N-19	Issue 6	S-11	AL 3				
N-21	Issue 6	E-1	AL 3				
N-23	AL 1	E-3	Issue 6				
N-25	Issue 6	E-5	AL 3				
N-27	Issue 6	E-7	AL 3				
N-29	AL 3	E-9	Issue 6				

ANA 1-9 INCORPORATED

ANA No	10	11	_	 -	
Location	N-11	N-13 E-17			

NOTES TO USERS

- 1. These Flight Reference Cards (FRC) are complementary to the Hawk T Mk 1/1A Aircrew Manual (AP 101B-4401-15). The same conventions are used, and amendment procedure is similar.
- 2. Checks marked * in the cards are applicable only to weapons capable aircraft, and ** to the T Mk 1A variant only.
- 3. Actions printed in bold face are those which should be completed from memory.
- 4. Urgency of the Need to Land. The following terms are used to give guidance but are not intended to be precise definitions nor preclude relevant airmanship actions such as performing a low speed handling check when the integrity of the airframe is suspect.
 - Land ASAP. Land at the nearest airfield with a runway suitable for a safe landing
 - Land as Soon as Practicable. Land at the nearest airfield where you can land safely and expect practical assistance for your particular aircraft type

Comments and suggestions should be forwarded using a F765X to the 1 Gp Hawk STANEVAL, RAF Leeming, DL7 9PE for forwarding to OC Handling Sqn, Boscombe Down, Salisbury, SP4 0JE. (Boscombe Down).

Iss 6 AL 3 Oct 17

APPROACHING THE AIRCRAFT CHECKS

1. Safe direction

Initial. **External**

- 2. Configuration correct, chocks in
- 3. Armament boards
- MASS safe
- 5. Pins fitted for pylon stores, no pools of fluid
- Extinguisher present

INITIAL CHECKS — FRONT COCKPIT

Before entering the cockpit, check the following:

- MDC firing handles and MDC
- 2. Safety pin central stowage. .6 pins
- 3. Landing gear selector DOWN button fully in, UP button not turned to emergency override
- 4. Battery switches Off
- 5. ** MCP coolant Off
- 6. * WCP All selectors OFF
- 7. * MASS LOCK SAFE, green flag up
- 8. * Stick top Safe
- 9. Oxygen selector ON, check contents
- PEC dust cover Remove and stow
- 11. Camera/GPS/Crash pad . . . Secure

EXTERNAL CHECKS

For Weapons External Checks see N-33 to N-34

Systematically check aircraft for signs of damage and leaks and for security of panels, doors and attachments. Check oleos for correct extension and tyres for excessive wear or cuts. Check that all covers, plugs, intake blanks and ground locks are removed. Check lifting surfaces and engine intakes are clear of ice.

Make the following specific checks:

- 1. Canopy, MDC and windscreen Clean and undamaged
- 2. MDC external handles Flush
- 3. Hydraulic fluid reservoirs. . . No 1 System, 4.5 4.6 ltrs No 2 System, 4.1 - 4.2 ltrs Refer to S-9/10 if outside limits
- 4. Hydraulic filter indicators (4). .Buttons not protruding
- 5. Right main wheel bay

No 1 flying controls

accumulator pressure. 72 to 80 bar Wheelbrakes pressure gauge 82 to 90 bar

Left main wheel bay

Flaps standby pressure 203 to 211 bar

No 2 flying controls

accumulator pressure. 72 to 80 bar

INITIAL CHECKS — REAR COCKPIT

		, G. H. 1.1
		Correctly fitted to ejection seat, MDC firing handle and MDC firing unit
2.	Command ejection	
	selector	Down and Off
3.	Ejection seat	For solo flight check apron fitted, straps and leg restraints secure, no obstructions and PEC cover fitted. For dual flight carry out Ejection Seat Checks (N-5)
4.	Anti-g lever	As required
	Engine start switch	
6.	Tailplane standby trim	Cover fully down
	Anti-skid switch	
8.	Standby UHF switch	MAIN
9.	Landing gear selector	Red button in, UP and DOWN buttons both out, UP button not turned to emergency override
10.	Flap selector	PUPIL
11.	* ISIS sight	OFF
12.	* WMP override switch	NORMAL
	* Stick top	
	Lighting switches	
	UHF aerial switch	
	Oxygen selector	
	Oxygen regulator	
	PEC dust cover	
	F700 bag	•
	MDC firing unit safety pin * Rear Gunsight Cover	

Rear, Ejec Seat

EJECTION SEAT CHECKS

EJE	ECTION SEAT CHECKS	
1.	Seat firing handle safety pin	Fitted through housing & firing
3.	Manual separation handle Automatic Deployment Unit:	Locked down, safety pin removed
6.	b. Mode BTRU static trip rod Command ejection pipe(s) Command/BTRU	AUTO/MANUAL as required Secured to cross beam Visually confirmed not disconnected
9.	BTRU capsule	disconnected, pip pin fitted Check operating altitude Clear of obstructions
	Scissor shackle Droque shackle	Closed and flat
13. 14.	Parachute container Drogue gun piston	Closed, 2 ties intact
15.	Top latch: a. Indicator spigot	Flush with, or slightly protruding from, plunger
40	b. Plunger	Flush with, or slightly recessed into, housing face
	a. Cable	Removed
18.	Emergency oxygen a. Contents	Full Fitted Horizontal
20.	Oxygen regulator Go forward mechanism PSP:	Selector to 100% (forward) Check HPRU operates & locks
22.		Connector in spring clip
	b. Parachute lift websc. Retention strapd. Sticker straps	
	1. 2. 3. 4. 5. 6. 7. 8. 9. 10. 11. 12. 13. 14. 15. 16.	safety pin

STRAPPING-IN CHECKS

Immediately before entering the cockpit:

1. MDC firing unit safety pin. . . Remove and stow

After entering the cockpit

- 2. Rudder pedals.....Lock disengaged, reach adjusted
- 4. External lights As required
- 6. PEC......Insert, select AM
- 7. PSP......Connect 8. Leg restraint linesConnect
- 9. Go forward mechanism. . . . Locked (rear)
- 10. Straps Route correctly, insert into QRF, tighten
- 11. Oxygen hose......Connected
- 12. MICTEL lead Connected and tethered
- 13. Go forward mechanism. . . . Check function
- 14. Seat HeightAs required
- 15. ** Radio (G/C intercom req). .V/UHF (main) tested
- 16. ** Ground crew intercom. . . Connected if required
- 17. CCS amplifier NORM
- 18. Intercom Set as required
- 19. Parking brake On

CANOPY CLOSING CHECKS

May be carried out after INTERNAL CHECKS if required

- 1. * Rear Gunsight Cover Check removed
- 2. Oxygen mask On
- 4. Groundcrew. Clear of aircraft
- 6. Harness......Secure, correctly routed
- 7. PSP and leg restraint lines. .Secure, correctly routed
- 8. Seat firing handle Clear of obstructions
- 9. Command ejection selector Confirmed Down and Off (if dual)
 - 10. MDC firing handle and
 - seat pins......Remove and stow
- 11. Command ejection selector As required
- 12. Rear cockpit switches Position confirmed (if dual)

	INTERNAL CHECKS - FRONT COCKPIT
	Left Console
Internal	1. LP cock
	 11. Anti-skid switch Off 12. Standby UHF switch MAIN 13. HYD 1 and 2 gauges Condition 14. Brake gauges Check pressures
	Left Instrument Panel
	 U/C standby handle Not pulled Flap standby handle Not pulled Landing gear indicator Three greens Voltmeter
	position if not down) 20. Flap indicator
	21. UHF
	Centre Instrument and Lower Panels
	28. DGI

37. 38. 39. 40. 41. 42. 43.	AHRS control unit * ISIS control unit	Condition Test NORMAL Zero, condition Condition, flag showing Erect, flag retracted
Rig	ht Instrument Panel	
46. 47.	Standby altimeter	Condition OXY caption out, Contents over ½ full Confirm AM, check flow. Select 100%, check flow. Reselect AM. Press test, check MI functioning.
	Fuel gauge	
	TGT gauge	
51. 52.	Rotation indicator	Black
	Lighting switches	As required
Rig	ht Console	
56.	CCS	As required. Set PTT to NORM. Check FAIL, reset NORM
57.	VHF	
	UHF aerial switch	•
	Cabin conditioning	NORMAL
	Cabin air temperature switch	AUTO sector, 6 o'clock approx
61. 62. 63. 64.	Ground intercom switch ADR status indicator IFF/SSR	As required FAIL SBY
66.	Radio(s)	Test as required

ENGINE STARTING

WARNING: Do not select Idle before rotation indicator green

Pre-sta	art
---------	-----

 Engine start switch (both cockpits) ON

2. Anti-collision lights As required

Starting

Eng Start

3. Relight button Press and release

When GTS indicator green

4. Engine start switch START and release

When rotation indicator green and RPM 15 to 20%

5. Throttle Idle

During Start

6. TGT and RPM Monitor. If TGT appears likely to exceed 570°C, set throttle to HP Off (Note 1)

7. At 45% RPM GTS and rotation indicators black

When RPM stabilized:

8. Fire warnings..... Out

10. TGT Note

11. OIL, FPR and TRANS

captions Out (Note 2)

12. Throttle Close bleed valve

13. Hydraulics Control response and feel normal on HYD 1 system

14. Reset HYD 2. Check pressure normal

15. Electrics........... GEN and AC captions out; press RESET buttons if

necessary.

Voltmeter 27 to 29 volts

16. CWP All captions out except SKID

NOTE 1: Under transient conditions 570°C may be exceeded by up to 20°C for 10 seconds

NOTE 2: FPR caption may flicker as booster pump is switched from Essential Services busbar to Generator busbar

START FAILURES

GTS

WARNING: Allow 3 minutes between the termination of one GTS start cycle and the commencement of the next. After three consecutive start cycles allow 20 minutes to elapse for DC motor cooling

No GTS Start (Switch Selection Error)

Check:

- 1. LP cock ON
- 2. Engine start switch

(both cockpits) ON

- 3. Ignition switch NORMAL
- 4. If DC motor has operated, wait 3 minutes then:
- 5. Relight button Press

No GTS Green (within 22 seconds maximum)

- 1. Allow start cycle to terminate
- 2. Report defect

Engine

No Rotation Green (within 10 seconds maximum)

- 1. Throttle Confirm HP Off
- 2. Engine start switch OFF
- Report defect

Engine Fails to Light (within 15 seconds of selecting Idle)

- 1. Throttle HP Off
- 2. Carry out Dry Crank below

Dry Crank

- 1. RPM Below 20%
- 2. GTS indicator Green (if not running allow 3 minutes since last cycle)
- 3. Ignition switch ISOLATE
- 4. Engine start switch START and release
- 5. Rotation indicator Green

When cycle complete:

- 6. Ignition switch NORMAL
- 7. Attempt further start

AP101B-4401-14

N-11

Iss 6 AL 3 Oct 17

AFTER START CHECKS

1. Trims Check trim functions, set TPI 0

2. Tailplane external position. . Confirm alignment 3. Anti-g..... Test

4. Airbrake..... Test

5. Pitot heat On

6. Flaps Check operation. Select MID

7. Controls..... Full, free and correct movement (when worn, FCAGTs should be fully inflated throughout check)

Fail start/ After start 8. Hydraulics Pressures within limits

9. Oxygen Contents, connections, flow 10. Canopy Locked, safety latch engaged,

arrows in line, canopy seal 11. Instruments (flight). Altimeters test, set QFE.

inflated All warning flags clear, erect, synchronize compass. Standby compass check,

DGI set, AHRS (post MOD

SIFF, confirm BIT complete

12. Instruments (navigation) . . IFF, set XMT, test,

ADR black, ILS as required, Tacan - test 13. *ISIS Test. Set as required

2516) check mode

14. **Ground crew intercom . . Disconnected **AIM-9L Acquisition Round

1. Nose protective cover.... Removed

IR Detection Test

2. MCP

a. Coolant switch TEST ON

b. Test indicator Red light on

c. Missile selected indicator.. Correct light on. If incorrect press reject button once

d. Volume Set as required

3. Growl..... Check

If second missile check required

4. Reject button. Press

5. Growl..... Check

After checks complete

6. MCP coolant switch As required

ANA 10

TAXYING CHECKS

1. Brakes Functional check 2. Instruments.... Functional check

BEFORE TAKE-OFF CHECKS

1. Trims Tailplane 0°, aileron and rudder neutral (see Note) 2. Pitot heat On

3. Anti-skid switch On. SKID caption out,

braking normal

4. Fuel Contents

5. Flaps MID

6. Harness..... Locked and tight.

7. Leg restraints and PSP... Connected

8. Weight Set

9. Visor Down

10. MDC firing handle and

seat pins..... Stowed 11. Central stowage 8 Pins

12. Command ejection

selector Position confirmed (if dual)

13. External lights As required

14. Take-off brief Complete

NOTE: Underwing stores/target tow/luneberg lens/forward CG -

Tailplane 1° nose up

Asymmetric stores - Aileron ½ and rudder ¼ deflection toward lighter wing Target tow - Rudder ⅓ deflection right

RUNWAY CHECKS

1.	IFF As Required
2.	Lights As Required
3.	Altimeters Within limits
4.	*MASS UNLOCK LIVE
5.	*WCP busbar indicators Black
6.	**WCP role indicator As required
7.	RPM
8.	TGT665°C maximum
9.	CWP Clear
10.	Wheelbrakes Holding

NOTE: New brakes may not hold at full power

AFTER TAKE-OFF CHECKS

1. Landing gear	Up below 200 knots
	UP button fully in
2. Flaps	Up below 200 knots
3 CWP	Clear

PRE-DESCENT CHECKS

1. Fuel Contents

2. Instruments Erect and synchronised

3. Radio..... Frequency set A/R

4. Altimeters Set A/R, cross checked

5. Demist A/R

6. Safety Altitude and safe height/altitude for descent

APPROACH DATA

Instrument Approach Settings

Pre T/O App Data

	Landing Gear	Flaps	Airspeed (knots)	RPM (%)
Downwind	UP	UP	230	80
Base Leg	DOWN	MID	160 / 150	85 to 87
Glidepath	DOWN	DOWN	160/150†	85 to 87
† Reducing steadily to 140 / 130 at 200 to 300 feet AGL				

Approach Category

Aircraft category for approaches is C

Circuit Speeds (knots)

N-13

	Circuit	Final Turn	Approach	Threshold
Powered Approach	190 reducing to 160 / 150†	160 / 150† reducing to 140 / 130†	140 / 130†	110 †† (Stores 115 ††)
PFL ANA 11	180 (Gear down / flap up 170 / 175†)	Gear down / flap up 170 / 175†		150(Min) 170 (Max) with down flap
Mid-flap Approach	190 reducing to 170 / 160†	170 / 160† reducing to 160 / 150†	160 / 150†	120 †† (Stores 125 ††)
Flapless Approach	190 reducing to 170 / 160†	170 / 160† reducing to 160 / 150†	160 / 150†	135 †† (Stores 140 ††)

† Use the higher speed if AUM is more than 5000 kg

ANA 11

†† Add 1 knot per 100 kg of fuel for all configurations

1. 2. 3.	FORE LANDING CHECK Speed	Below 200 knots IN, MI black DOWN, button fully in
5. 6. 7. 8.	Harness Landing gear	Set and indicated as required Locked and tight
Sho	ort Final	1171
1. 2.	Configuration	Gear DOWN, flaps A/R Toes Clear
	TER LANDING CHECKS	
2.	*MASS *MASS key Command ejection	
	selector	
6. 7. 8. 9. 10. 11.	Pitot heat	Off All pressures normal As required Safe OFF OFF
14. 15. 16. 17. 18. 19.	indicators	OFF Test As required As required OFF Minimum 6 o'clock position
21. 22.	ILS Tacan	OFF
1.	*Cockpit (if appropriate) *Override weapons switch . *Sticktop	

Landing Shutdown

SHUTDOWN CHECKS - FRONT COCKPIT

Note: Before shutting down the engine allow 30 seconds at Idle for RPM and TGT to stabilize. Do not switch fuel pump off until just before setting throttle to HP Off

1. Parking brake On
2. Engine start switch OFF
3. Fuel pump switch Off
4. Throttle
5. Oxygen mask Donned
6. Visors Down
7. Canopy Open
8. Ejection seat
9. Shoulder straps Stow on headbox
Note: When unstrapped, return QRF to locked position
10. LP Fuel cockOff
11. RadioOFF
12. *ISIS OFF
13. External lightsOff
14. Cockpit lighting OFF
15. Parking brake As required
Note: With the parking brake left on system pressure will dissipate
with time
16. Hydraulics Control response and feel normal on HYD 2 system

- 17. Battery switches OFF
- 18. PEC dust cover........Fit to seat
- 19. Oxygen selector OFF
- 20. Oxygen regulator 100%
- 21. MDC firing unit safety pin Fit
- 22. Seat firing handle

SHUTDOWN CHECKS - REAR COCKPIT

Note: The engine is to be shutdown prior to commencing

- 1. Ejection seat Unstrap
- 2. Shoulder straps Stow on headbox

Note: When unstrapped, return QRF to locked position

- 3. Engine start switch OFF
- 4. Anti-skid switch Off
- 5. Cockpit lighting OFF
- 6. PEC dust cover.....Fitted
- 7. Oxygen selector OFF
- 8. Oxygen regulator 100%
- 9. MDC firing unit safety pin . . . Fit

Seat firing handle

safety pin Confirm fitted through housing & firing handle

Before leaving the aircraft confirm the aircraft is Safe for Parking

DI 0 - PLANNING DATA

	Normal (Climb - 1200l	kg	Light Climb - 500kg			
FL	Distance	Time	Fuel	Distance	Time	Fuel	
100	10	1:36	40	8	1:24	34	
200	26	3:48	86	22	3:12	73	
250	34	5:00	106	29	4:12	89	
300	45	6:24	126	37	5:24	106	
350	59	8:24	149	48	6:54	124	
400	83	11:54	182	65	9:24	147	

Climb data is based on a climb from sea level, at climb speed, ISA and still wind. Tables do not include 60kg for SUTTO. Time mm:ss.

Cruise Data

ML Cruise - (0.75M, 5000kgs)								
FL	Kg / min	Kg / Anm						
100	18.5	2.32						
200	12.2	1.59						
250	10	1.32						
300	8.4	1.14						
350	6.7	0.93						
400	6.0	0.84						

LL Cruise	
300 kts	10.5 kg/min
360 kts	13.7 kg/min
420 kts	18.9 kg/min
480 kts	25.9 kg/min

Descent Data

		nstrumen	t	Range			
FL	Distance	Time	Fuel	Distance	Time	Fuel	
450	41	6:36	30	83	13:00	32	
400	34	5:24	27	69	11:00	28	
350	28	4:30	24	57	9:06	24	
300	23	3:48	21	47	7:42	21	
250	19	3:12	19	39	6:30	18	
200	15	2:36	16	31	5:18	15	
100	7	1:18	9	14	2:36	8	
		Tactical		Nav			
450	17	2:24	6	58	7:48	22	
400	14	1:00	5	46	6:18	18	
350	11	1:30	4	37	5:00	15	
300	9	1:15	4	30	4:12	13	
250	3	1:06	7	25	3:30	11	
200	6	0:54	3	22	3:00	10	
100	3	0:30	1	10	1:30	5	

RECOVERY DATA - DI 0

- 1. Assume ISA & Still Air.
- 2. Range and endurance figures are to 200 kg at chosen altitude.
- 3. Range includes climb and cruise but not descent distance
- 4. M = Maintain altitude, † = Fuel remaining after engine start, taxy, take-off, accelerate to climb speed and climb

DI 0 - Sea Level									
Fuel kg	†1212	800	600	500	400	300			
Maintain Sea Level (95% R: 280 to 295kts)									
R: 230	528 nm	317 nm	213 nm	160 nm	107 nm	54 nm			
E: 155-170	168 min	103 min	70 min	53 min	36 min	18 min			
ANM/100kg	51	52	53	53	53	53			
kg/nm	1.96	1.92	1.89	1.89	1.89	1.89			
	Cli	mb to Op	timum FL	(350/0.73)				
FL	400	400	400	350/400	350	50/100			
Mach	0.7	0.7	0.67	0.63/0.66	0.63	0.36/0.38			
Range	1076 nm	618 nm	382 nm	261 nm	144 nm	55 nm			

Recovery DI 0

DI 0 - FL100									
Fuel kg	†1172	800	600	500	400	300			
Maintain FL100 (95% R: 275 to 295kts)									
R: 230	649 nm	407 nm	274 nm	206 nm	138 nm	69 nm			
E: 155-170	183 min	117 min	79min	60 min	40 min	20 min			
ANM/100kg	64	66	67	68	68	69			
kg/nm	1.56	1.52	1.49	1.47	1.47	1.45			
	Cli	mb to Op	timum FL	(350/0.73)				
FL	400	400	400	400	350	150			
Mach	0.7	0.7	0.67	0.66	0.63	0.43			
Range	1066 nm	651 nm	416 nm	294 nm	175 nm	70 nm			

DI 0 - FL200										
Fuel kg	†1126	800	600	500	400	300				
	Maintain FL200 (95% R: 255 to 290kts)									
R: 210	787 nm	517 nm	348 nm	262 nm	175 nm	88 nm				
E: 155-170	188 min	127 min	86 min	65 min	44 min	22 min				
ANM/100kg	81	84	85	86	87	88				
kg/nm	1.24	1.19	1.18	1.16	1.15	1.14				
	Climb to Optimum FL (350/0.73)									
FL	400	400	400	400	350	250				
Mach	0.7	0.7	0.67	0.66	0.63	0.51				
Range	1050 nm	686 nm	450 nm	329 nm	207 nm	91 nm				

RECOVERY DATA - DI 0 - continued

DI 0 - FL300										
Fuel kg	†1086	800	600	500	400	300				
	Maintain FL300 (95% R: 0.68 to 0.71)									
R: 0.6	934 nm	642 nm	433 nm	326 nm	219 nm	110 nm				
E: 160-180	184 min	129 min	88 min	67 min	45 min	23 min				
ANM/100kg	100	103	106	107	108	109				
kg/nm	1	.97	.94	.93	.93	.92				
	Cli	mb to Op	timum FL	(350/0.73	3)					
FL	400	400	400	400	350/400	350				
Mach	0.7	0.7	0.67	0.66	0.63/0.65	0.63				
Range	1031 nm	711 nm	476 nm	354 nm	231 nm	111 nm				

DI 0 - FL350										
Fuel kg	†1062	800	600	500	400	300				
	Maintain FL350 (95% R: 0.72 to 0.73)									
R: 0.65	995 nm	705 nm	476 nm	359 nm	241 nm	121 nm				
E: 170-185	179 min	129 min	88	67 min	45 min	23 min				
ANM/100kg	109	113	116	117	119	120				
kg/nm	.92	.88	.86	.85	.84	.83				
	Climb to Optimum FL (350/0.73)									
FL	400	400	400	400	400	М				
Mach	0.7	0.7	0.67	0.66	0.65	0.63				
Range	1016 nm	722 nm	487 nm	366 nm	242 nm	121 nm				

DI 0 - FL400										
Fuel kg	†1029	800	600	500	400	300				
	Maintain FL400 (95% R: 0.74 to 0.76)									
R: 0.65-0.7	992 nm	734 nm	499 nm	378 nm	254 nm	128 nm				
E: N/A	Descend to FL300 to FL350									
ANM/100kg	110	115	120	122	125	127				
kg/nm	.91	.87	.83	.82	.80	.79				
	Cli	mb to Op	timum FL	(350/0.73)					
FL			Maintair	n FL400						
Mach	0.7	0.7	0.67	0.66	0.65	0.65				
Range	992 nm	734 nm	499 nm	378 nm	254 nm	128 nm				

Aug 12

RECOVERY DATA - DI 0 - continued

DI 0 - FL450									
Fuel kg	†939	800	600	500	400	300			
	Maintain FL450 (95% R: 0.75 to 0.76)								
R: 0.7	899 nm	744 nm	509 nm	387 nm	261 nm	132 nm			
E: N/A		Descend to FL300 to FL350							
ANM/100kg	109	114	121	124	127	131			
kg/nm	.92	.88	.87	.81	.79	.76			
	Cli	mb to Op	timum FL	(350/0.73)				
FL			Maintair	n FL450					
Mach	0.7	0.7	0.7	0.7	0.7	0.7			
Range	899 nm	744 nm	509 nm	387 nm	261 nm	132 nm			

Endurance - Clean

Sea level 15.5 minutes/100 kg at 700 kg remaining 20,000 ft 18.5 minutes/100 kg at 700 kg remaining 30,000 to 35,000 ft 18.5 minutes/100 kg at 700 kg remaining From low level with less than 600kg do not climb above 10,000 ft

DI 5 - PLANNING DATA

	Normal Climb - 1200kg			Light Climb - 500kg		
FL	Distance	Time	Fuel	Distance	Time	Fuel
100	9	1:42	42	8	1:30	36
200	23	4:00	89	20	3:24	75
250	32	5:00	109	26	4:24	91
300	42	7:00	131	35	5:48	109
350	58	9:00	158	47	6:36	129
400	89	14:00	202	66	9:42	156

Climb data is based on a climb from sea level, at climb speed, ISA and still wind. Tables do not include 60Kg for SUTTO, Time mm:ss.

Cruise Data

ML Cruise - (0.70M, 5000kgs)									
FL	Kg / min	Kg / Anm							
100	17.2	2.31							
200	11.3	1.58							
250	9.3	1.32							
300	7.9	1.15							
350	7.1	1.06							
400	7.0	1.05							

LL Cruise					
300 kts	11.4 kg/min				
360 kts	15.1 kg/min				
420 kts	20.8 kg/min				
480 kts	28.5 kg/min				

Descent Data

	Instrument			Range		
FL	Distance	Time	Fuel	Distance	Time	Fuel
400	33	5:15	26	63	10:00	26
350	27	4:24	23	52	8:12	22
300	22	3:42	21	43	7;00	19
250	18	3:06	18	36	6:00	17
200	15	2:36	16	29	4:54	14
100	7	1:18	9	13	2:24	7
		Tactical		Nav		
400	16	2:12	6	42	5:42	18
350	12	1:42	5	34	4:42	14
300	10	1:24	4	27	3:54	12
250	8	1:06	3	23	3:12	10
200	6	0:54	3	19	2:42	8
100	3	0:30	1	9	1:24	4

RECOVERY DATA - DI 5

- 1. Assume ISA & Still Air.
- 2. Range and endurance figures are to 200 kg at chosen altitude.
- 3. Range includes climb but not descent distance
- 4. M = Maintain altitude, † = Fuel remaining after engine start, taxy, take-off, accelerate to climb speed and climb

	DI 5 - Sea Level								
Fuel kg	†1212	800	600	500	400	300			
IV	Maintain Sea Level (95% R: 270 to 290kts)								
R: 230	490 nm	294 nm	197 nm	149 nm	99 nm	50 nm			
E: 155-170	156 min	96 min	65 min	49 min	33 min	17 min			
ANM/100kg	47	48	49	49	49	50			
kg/nm	2.13	2.08	2.04	2.04	2.04	2.00			
	Climb to Optimum FL (300/0.65)								
FL	350	350	350	350	300	M/50			
Mach	0.64	0.64	0.63	0.63	0.63	0.35			
Range	941 nm	540 nm	337 nm	233 nm	128 nm	50 nm			

Recovery DI 5

	DI 5 - FL100								
Fuel kg	†1170	800	600	500	400	300			
	Maintain FL100 (95% R: 255 to 290kts)								
R: 210	594 nm	374 nm	251 nm	189 nm	127 nm	64 nm			
E: 155-170	171 min	116 min	79min	60 min	40 min	20 min			
ANM/100kg	59	61	62	62	63	63			
kg/nm	1.69	1.64	1.61	1.61	1.59	1.59			
	Cli	mb to Op	timum FL	(300/0.65)				
FL	350	350	350	350	350	M/150			
Mach	0.64	0.64	0.63	0.63	0.63	0.37/0.43			
Range	917 nm	601 nm	397 nm	292 nm	186 nm	82 nm			

	DI 5 - FL200								
Fuel kg	†1123	800	600	500	400	300			
	Mainta	in FL200	(95% R:	260 to 280	Okts)				
R: 230	710 nm	469 nm	316 nm	238 nm	159 nm	80 nm			
E: 155-175	171 min	116 min	79 min	60 min	40 min	20 min			
ANM/100kg	74	76	77	78	79	80			
kg/nm	1.35	1.32	1.30	1.28	1.27	1.25			
	Climb to Optimum FL (300/0.65)								
FL	350	350	350	350	350	250/300			
Mach	0.64	0.64	0.63	0.63	0.63	0.53/0.58			
Range	917 nm	601 nm	397 nm	292 nm	186 nm	82 nm			

RECOVERY DATA - DI 5 - continued

	DI 5 - FL300							
Fuel kg	†1081	800	600	500	400	300		
	Main	tain FL30	0 (95% R	: 0.67 to 0).7)			
R: 0.6	841 nm	581 nm	391 nm	295 nm	198 nm	99 nm		
E: 165-185	165 min	116 min	79 min	60 min	40 min	20 min		
ANM/100kg	91	94	96	97	98	99		
kg/nm	1.10	1.06	1.04	1.03	1.02	1.01		
	Cli	mb to Op	timum FL	(300/0.65)			
FL	350	350	350	350	350	M/350		
Mach	0.64	0.64	0.63	0.63	0.63	0.58/0.63		
Range	898 nm	622 nm	418 nm	313 nm	207 nm	99 nm		

DI 5 - FL350								
Fuel kg	†1054	800	600	500	400	300		
	Maintain FL350 (95% R: 0.7 to 0.72)							
R: 0.63	883 nm	632 nm	428 nm	323 nm	217 nm	109 nm		
E: N/A	Descend to FL200 to FL350							
ANM/100kg	96	101	104	106	107	108		
kg/nm	1.04	.99	0.96	.94	.93	.93		
	Cli	mb to Op	timum FL	(300/0.65)			
FL			Maintair	n FL350				
Mach	0.64	0.64	0.63	0.63	0.63	0.63		
Range	883 nm	632 nm	428 nm	323 nm	217 nm	109 nm		

	DI 5 - FL400							
Fuel kg	†1009	800	600	500	400	300		
	Maintain FL400 (95% R: 0.73 to 0.75)							
R: 0.67	848 nm	642 nm	437 nm	331 nm	223 nm	113 nm		
E: N/A		Descend to FL200 to FL300						
ANM/100kg	97	101	105	107	109	111		
kg/nm	1.03	1	.95	.93	.92	,90		
	Cli	mb to Op	timum FL	(300/0.65)			
FL			Maintaiı	n FL400				
Mach	0.7	0.69	0.68	0.67	0.67	0.66		
Range	848 nm	642 nm	437 nm	331 nm	223 nm	113 nm		

Endurance - DI 5

Sea level 15.5 minutes/100 kg at 700 kg remaining 20,000 ft 18.5 minutes/100 kg at 700 kg remaining 30,000 to 35,000 ft 18.5 minutes/100 kg at 700 kg remaining From low level with less than 600kg do not climb above10,000 ft

DI 10 - PLANNING DATA

	Normal C	limb - 1200kg	g	Light Climb - 500kg		
FL	Distance	Time	Fuel	Distance	Time	Fuel
100	10	1:54	47	8	1:36	40
200	26	4:30	100	22	3:48	54
250	36	6:00	123	30	5:00	103
300	49	8:00	149	40	6:36	124
350	68	11:00	182	54	8:48	148
400	114	18:30	248	79	12:42	183

Climb data is based on a climb from sea level, at climb speed, ISA and still wind. Tables do not include 60 Kg for SUTTO, Time mm:ss.

Cruise Data

ML Cruise - (0.70M, 5000kgs)								
FL	Kg / min	Kg / Anm						
100	15.8	2.28						
200	10.6	1.59						
250	8.8	1.35						
300	7.6	1.19						
350	6.1	0.98						
400	5.8	0.93						

LL Cruise	
300 kts	12.4 kg/min
360 kts	16.6 kg/min
420 kts	22.6 kg/min
480 kts	31.1 kg/min

Recovery DI 10

Descent Data

	Instrument			Range			
FL	Distance	Time	Fuel	Distance	Time	Fuel	
400	32	5:12	25	59	9:15	24	
350	26	4:18	22	48	7:42	21	
300	21	3:36	20	40	6:30	18	
250	18	3:00	18	33	5:36	16	
200	14	2:30	15	26	4:30	15	
100	7	1:18	8	12	2:12	6	
		Tactical		Nav			
400	15	2:06	15	39	5:18	16	
350	12	1:42	12	31	4:18	13	
300	9	1:18	9	25	3:36	11	
250	7	1:06	7	21	3:00	9	
200	6	0:54	6	18	2:30	7	
100	3	0:30	3	8	1:18	4	

RECOVERY DATA - DI 10

- 1. Assume ISA & Still Air.
- 2. Range and endurance figures are to 200 kg at chosen altitude.
- 3. Range includes climb but not descent distance
- 4. M = Maintain altitude, † = Fuel remaining after engine start, taxy, take-off, accelerate to climb speed and climb

DI 10 - Sea Level									
Fuel kg	†1212	800	600	500	400	300			
	Maintain	Sea Leve	el (95% R	: 260 to 2	80kts)				
R: 220	460 nm	276 nm	186 nm	140 nm	93 nm	47 nm			
E: 155-175	148 min	91 min	62 min	47 min	31 min	16 min			
ANM/100kg	44	45	46	46	46	47			
kg/nm	2.27	2.22	2.17	2.17	2.17	2.13			
	Climb to Optimum FL (300/0.65)								
FL	350	350	350	350	250/300	M/50			
Mach	0.63	0.63	0.62	0.62	0.51/0.57	0.33/0.35			
Range	849 nm	485 nm	300 nm	205 nm	112 nm	47 nm			

DI 10 - FL100							
Fuel kg	†1165	800	600	500	400	300	
	Mainta	in FL100	(95% R:	250 to 280	Okts)		
R: 210	552 nm	349 nm	235 nm	177 nm	118 nm	60 nm	
E: 155-175	157 min	101 min	69 min	52 min	35 min	18 min	
ANM/100kg	55	56	57	58	59	59	
kg/nm	1.82	1.79	1.75	1.72	1.69	1.69	
	Cli	mb to Op	timum FL	(300/0.65)		
FL	350	350	350	350	300	М	
Mach	0.63	0.63	0.62	0.62	0.57	0.36	
Range	839 nm	515 nm	329 nm	265 nm	139 nm	60 nm	

DI 10 - FL200									
Fuel kg	†1112	300							
	Maintain FL200 (95% R: 250 to 280kts)								
R: 210	648 nm	433 nm	292 nm	220 nm	148 nm	74 nm			
E: 160-175	158 min	108 min	74 min	56 min	38 min	19 min			
ANM/100kg	68	70	71	72	73	74			
kg/nm	1.47	1.43	1.41	1.39	1.37	1.35			
	Climb to Optimum FL (300/0.65)								
FL	350	350	350	350	350	250			
Mach	0.63	0.63	0.62	0.62	0.62	0.51			
Range	822 nm	544 nm	359 nm	264 nm	167 nm	75 nm			

Aug 12

RECOVERY DATA - DI 10 - continued

DI 10 - FL300							
Fuel kg	†1062	800	600	500	400	300	
	Maint	ain FL300	(95% R:	0.66 to 0	.68)		
R: 0.58	761 nm	537 nm	362 nm	273 nm	183 nm	92 nm	
E: 165-185	151 min	108 min	73 min	55 min	37 min	19 min	
ANM/100kg	84	87	88	89	90	91	
kg/nm	1.19	1.15	1.14	1.12	1.11	1.09	
	Cli	mb to Op	timum FL	(300/0.65)		
FL	350	350	350	350	350	М	
Mach	0.63	0.63	0.62	0.62	0.62	0.57	
Range	800 nm	565 nm	380 nm	285 nm	188 nm	92 nm	

DI 10 - FL350							
Fuel kg	†1029	800	600	500	400	300	
	Maint	ain FL350) (95% R:	0.68 to 0	.71)		
R: 0.62	781 nm	575 nm	390 nm	294 nm	198 nm	100 nm	
E: N/A		Des	cend to Fl	_200 to FL	300		
ANM/100kg	88	91	94	96	97	99	
kg/nm	1.14	1.09	1.06	1.04	1.03	1.01	
	Cli	mb to Op	timum FL	(300/0.65)		
FL			Maintaiı	n FL350			
Mach	0.63	0.63	0.62	0.62	0.62	0.61	
Range	781 nm	575 nm	390 nm	294 nm	198 nm	100 nm	

DI 10 - FL400							
Fuel kg	†962	800	600	500	400	300	
	Maint	ain FL400	(95% R:	0.72 to 0	.74)		
R: 0.66	726 nm	581 nm	396 nm	300 nm	202 nm	102 nm	
E: N/A		Descend to FL200 to FL300					
ANM/100kg	88	91	95	97	99	-	
kg/nm	1.14	1.09	1.05	1.03	1.01		
	Cli	mb to Op	timum FL	(300/0.65)		
FL			Maintair	n FL400			
Mach	0.68	0.67	0.66	0.65	0.65	0.64	
Range	726 nm	581 nm	396 nm	300 nm	202 nm	102 nm	

Endurance - DI 10

Sea level 14.7 minutes/100 kg at 700 kg remaining 20,000 ft 17.3 minutes/100 kg at 700 kg remaining 30,000 to 35,000 ft 17.2 minutes/100 kg at 700 kg remaining From low level with less than 600kg do not climb above10,000 ft

DI 15 - PLANNING DATA

	Normal Climb - 1200kg			Light Climb - 500kg		
FL	Distance	Time	Fuel	Distance	Time	Fuel
100	11	2:06	52	10	1:48	45
200	31	5:18	115	26	4:24	97
250	43	7:06	144	36	5:54	121
300	59	9:36	178	48	7:54	147
350	85	13:42	222	66	10:48	178
400	180	29:00	354	104	16:48	231

Climb data is based on a climb from sea level, at climb speed, ISA and still wind. Tables do not include 60 Kg for SUTTO, Time mm:ss.

Cruise Data

ML Cruise - (0.65M, 5000kgs)								
FL	Kg / min	Kg / Anm						
100	17.1	2.47						
200	11.4	1.71						
250	9.4	1.44						
300	8.3	1.30						
350	7.6	1.22						
400	7.6	1.22						

LL Cruise	
300 kts	13.3 kg/min
360 kts	18.1 kg/min
420 kts	24.5 kg/min
480 kts	34.0 kg/min

Descent Data

	Instrument				Range		
FL	Distance	Time	Fuel	Distance	Time	Fuel	
400	31	5:00	25	54	8:30	22	
350	25	4:12	22	44	7:06	19	
300	21	3:30	20	36	6:00	17	
250	18	3:00	18	30	5:06	15	
200	14	2:30	15	24	4:12	12	
100	7	1:18	8	11	2:00	6	
		Tactical		Nav			
400	15	2:06	5	36	4:54	14	
350	12	1:42	4	29	4:00	12	
300	9	1:18	4	24	3:24	10	
250	7	1:06	3	20	2:54	9	
200	6	0:54	3	18	2:24	7	
100	3	0:30	1	7	1:12	4	

RECOVERY DATA - DI 15

- 1. Assume ISA & Still Air.
- 2. Range and endurance figures are to 200 kg at chosen altitude.
- 3. Range includes climb but not descent distance
- 4. M = Maintain altitude, † = Fuel remaining after engine start, taxy, take-off, accelerate to climb speed and climb

	DI 15 - Sea Level						
Fuel kg	†1212	800	600	500	400	300	
	Maintain	Sea Leve	el (95% R	: 255 to 2	75kts)		
R: 220 430 nm 259 nm 174 nm 131 nm 88 nm 44 n					44 nm		
E: 160-180	139min	85 min	58 min	44 min	29 min	15 min	
ANM/100kg	41	42	43	43	43	44	
kg/nm	2.44	2.38	2.33	2.33	2.33	2.27	
	Climb to Optimum FL (300/0.65)						
FL	350	350	350	300	150	М	
Mach	0.63	0.62	0.61	0.57	0.41	0.32	
Range	756 nm	428 nm	261 nm	178 nm	98 nm	44 nm	

DI 15 - FL100							
Fuel kg	†1159	800	600	500	400	300	
	Mainta	in FL100	(95% R:	245 to 270	Okts)		
R: 210	5102 nm	324 nm	218 nm	165 nm	110 nm	55 nm	
E: 160-180	145 min	94 min	64 min	48 min	33 min	16 min	
ANM/100kg	51	52	53	54	55	55	
kg/nm	1.96	1.92	1.89	1.85	1.82	1.82	
	Climb to Optimum FL (300/0.65)						
FL	350	350	350	300/350	300	М	
Mach	0.63	0.62	0.61	0.57/0.61	0.56	0.36	
Range	744 nm	457 nm	290 nm	204 nm	122 nm	55 nm	

DI 15 - FL200							
Fuel kg	†1097	800	600	500	400	300	
	Mainta	in FL200	(95% R:	250 to 280	Okts)		
R: 215	588 nm	399 nm	269 nm	203 nm	136 nm	68 nm	
E: 160-180	143 min	99 min	68 min	51 min	35 min	18 min	
ANM/100kg	63	64	66	66	67	68	
kg/nm	1.59	1.56	1.52	1.52	1.49	1.47	
	Climb to Optimum FL (300/0.65)						
FL	350	350	350	350	300	M/250	
Mach	0.63	0.62	0.61	0.61	0.56	0.44/0.51	
Range	725 nm	486 nm	320 nm	234 nm	149 nm	68 nm	

Recovery DI 15

RECOVERY DATA - DI 15 - continued

DI 15 - FL300							
Fuel kg	†1034	800	600	500	400	300	
	Maint	ain FL300	(95% R:	0.65 to 0	.67)		
R: 0.57	676 nm	493 nm	333 nm	251 nm	168 nm	85 nm	
E: 170-190	133 min	99 min	67 min	51 min	34 min	17 min	
ANM/100kg	77	79	81	82	83	84	
kg/nm	1.30	1.27	1.24	1.22	1.20	1.19	
	Climb to Optimum FL (300/0.65)						
FL	350	350	350	350	M/350	М	
Mach	0.63	0.62	0.61	0.61	0.56/0.61	0.56	
Range	696 nm	507 nm	341 nm	255 nm	168 nm	85 nm	

DI 15 - FL350						
Fuel kg	†989	800	600	500	400	300
	Maint	ain FL350	(95% R:	0.68 to 0	.71)	
R: 0.61	670 nm	517 nm	350 nm	265 nm	178 nm	90 nm
E: N/A		Des	cend to Fl	_250 to FL	.300	
ANM/100kg	80	82	85	86	88	89
kg/nm	1.25	1.22	1/18	1.16	1.14	1.12
	Climb to Optimum FL (300/0.65)					
FL	Maintain FL350					
Mach	0.63	0.61	0.61	0.61	0.61	0.60
Range	670 nm	517 nm	350 nm	265 nm	178 nm	90 nm

DI 15 - FL400						
Fuel kg	†854	800	600	500	400	300
	Maint	ain FL400) (95% R:	0.72 to 0	.73)	
R: 0.64	561 nm	436 nm	352 nm	267 nm	180 nm	91 nm
E: N/A		Des	cend to Fl	_200 to FL	300	
ANM/100kg	80	82	84	86	88	90
kg/nm	1.25	1.22	1.19	1.16	1.14	1.11
	Climb to Optimum FL (300/0.65)					
FL	Maintain FL400					
Mach	0.66	0.65	0.64	0.64	0.64	0.63
Range	561 nm	436 nm	352 nm	267 nm	180 nm	91 nm

Endurance - DI 15

Sea level 13.7 minutes/100 kg at 700 kg remaining 20,000 ft 15.8 minutes/100 kg at 700 kg remaining 30,000 to 35,000 ft 15.8 minutes/100 kg at 700 kg remaining From low level with less than 600kg do not climb above10,000 ft

Lims

LIMITATIONS

NOTE: The following limitations are taken from the MOD (AFD) Release document, which should be consulted for the latest release standard

Airframe

BASIC AIRCRAFT WITH OR WITHOUT GUNPOD

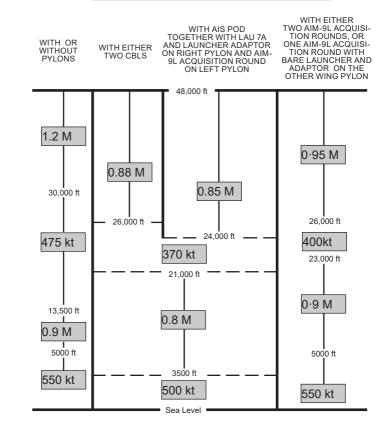
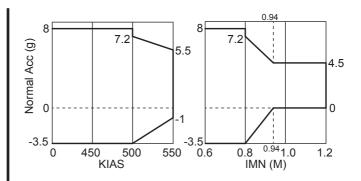


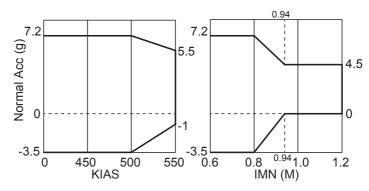
Fig 1 Airspeed, Mach Number and Altitude Limits

Landing gear (locked down or cycling) 200 knots maximum Flaps (extended or cycling) 200 knots maximum

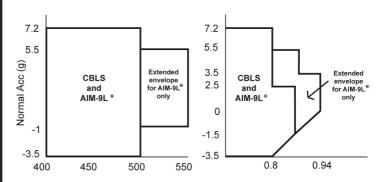
LIMITATIONS - continued



BASIC AIRCRAFT NORMAL ACCELERATION LIMITS



GUNPOD AND/OR BARE PYLON NORMAL ACCELERATION LIMITS



*AIM-9L AQUISTION ROUND, AIS POD or CBLS NORMAL ACCELERATION LIMITS

Fig 2 Normal Acceleration Limits (Flaps Up)

LIMITATIONS - continued

Normal Acceleration with Airbrake Extended

Normal Acceleration with Flaps Extended

Up to 200 knots Zero to +2.5g

Zero and Negative g Manoeuvres

Flight at zero or negative g conditions is not to exceed 30 seconds. A minimum interval of 10 seconds must be allowed between successive periods of zero or negative normal acceleration. The interval is to be allowed from either the restoration of positive g or the extinguishing of the CWP FUEL and OIL captions, whichever is the latter

Banner Target on Tow

Maximum speed

(except in emergency) 250 knots

Manoeuvres:

a. Angle of bank 45° maximum b. Acceleration +1.5g maximum

.....+0·5g minimum

c. Gentle rolling manoeuvres only

Weight

Maximum for take-off 5700 kg Maximum normal for landing . . 5000 kg

For braked landing above 4500 kg, the wheelbrake restrictions must be observed

Arresting Gear Trampling Speeds

Condition	Single or Mul	tiple Aircraft	Single Aircraft
Cable Type	Tensioned or unsupported	Tied down Bowspring High/Low	Supported †
RHAG	Any speed	Any speed	Any speed
PUAG	Any speed	-	Any speed

**

[†] Succeeding aircraft must not trample these cables until cable bounce has ceased

LIMITATIONS - continued

Standby Pressure Instrument Allowance +100 feet

Crosswind

Take-off and landing. 30 knots

Take-off and landing with asymmetric stores on downwind wing:

AIM-9L Acquisition Round 15 knots

Engine

Power Condition	RPM (%)	TGT (°C)	Time Limit
During Starting (Max)		570 (Note 3)	Unrestricted
During Re-lighting (Max)		585 (Note3)	Unrestricted
Idle (Nominal) (Bleed Valve Closed)	55.0 at ISA (Note 1)	450	Unrestricted
Maximum Continuous	99.3	615	Unrestricted
Maximum Power	104.0	665 (Note 2)	30 minutes per flight
Transient	-	685	20 sec per transient

NOTE 1: This is a nominal figure and varies slightly depending upon loading, air off-take and ambient conditions

NOTE 2: The TGT limiter is set to 660 ±5°C

NOTE 3: Under transient conditions the starting and relighting max TGTs may be exceeded by up to 20°C for 10 seconds

Fuel

NATO Code	UK Desig	UK Specification	US Desig
	А	PPROVED FUELS	
F-34 F-40 F-44	Avtur/FSII Avtag/FSII Avcat/FSII	91-87 Excluding Annex C 91-88 Issue 4 91-86 Issue 6	JP - 8 JP - 4 JP - 5
	AL	TERNATIVE FUELS	
F-35	Avtur	91-91 Excluding Annex D	JET A-1
F-24	Avtur/FSII	91-91 Excluding Annex D	JET A with Additives
	Avtur		JET A
			JET B

NOTE: For a full list of fuel limitations refer to the RTS

loil

The following engine oil is approved: OX27, NATO Code 0-156

* WEAPONS EXTERNAL CHECKS

Safety Checks

External power Removed MASS Green flag up

**

CBLS

Pylon ERU Armed (grey)

CBLS:

Carrier Secure on pylon

ERU Cocked, yellow bands visible

Electrical supplies Connected Access panel Secure

3 kg practice bomb:

Crutching Check security of bomb on

carrier

Nose cap ... Condition
Body fins ... Condition
Safety pin ... Removed

**

When practice bomb checks complete:
Pylon ERU safety pins Remove

Weapons

Weapons External Checks - continued

**

**AIM-9L	Acquisition	Round
----------	-------------	-------

Pylon ERU Unarmed (green)

Pylon ERU safety pins Remove

Missile Secure, clean, undamaged

1. Nose protective cover Remove, check head clean,

replace

2. GGG exhaust No carbon

3. Umbilical cord Connected

4. Detent hold-down pin Fitted, secure

5. LAU 7A detent

wrench safety pin. Removed

cover closed

**AIS Pod

Pylon ERU Unarmed (green)

Pylon ERU safety pins Remove

Pod Secure, clean, undamaged

1. Umbilical cord Connected

2. Detent hold-down pin Fitted, secure

3. LAU 7A detent

wrench safety pin Removed

After Flight

**AIM-9L Acquisition Round:

Nose protective cover. Fitted Pylon ERU safety pins Fitted

**AIS Pod

Pylon ERU safety pins Fitted

HAVEQUICK II PROCEDURES

Power up post-Mod 1015 select: MAIN. MANUAL Frequency display to 220.000 Preset channel 20, PRESET, check single beep Open access cover Press PRESET button, check single beep is heard Close access cover

HQ II Training Frequency (FMT) Load

Manual/Preset PRESET

Channel Selector Chan 20 (Beep)

Frequency Selectors 220.075

Preset Button (Red) Press and release (Beep)

Manual/Preset MANUAL Channel Selector Chan 20

Frequency Selectors Enter first training frequency Tone Button Press and release (Beep)

FMT 1

FMT 2

FMT 4

Repeat last 3 steps with decreasing channels until all 16 frequencies loaded (Chan 5 contains last frequency).

NATO Training Frequencies (FMT) FMT 3

FMT

Country	UK, GE, DE,SL, RO,BU	NO,FR, PO,GR, ES,LA,LI	IT,SP,BE, NL, LU,TU	CZ, HU, POL
20	252.725	261.050	252.925	252.100
19	374.125	379·325	374.425	373.800
18	265.875	269·350	264.550	257.500
17	315.875	316.550	308.550	310.000
16	284.950	291·250	283.875	280.325
15	357.150	359·350	344.925	355.675
14	342.575	338-950	337.025	344.525
13	363.275	373.875	357.325	363.700
12	387.850	399·450	386.800	399.775
11	379.225	386·550	379·725	378.025
10	292.200	306.500	292·425	291.175
9	298.575	310.900	300.725	298.275
8	248.275	253.000	245.725	246.775
7	336.025	328·400	312-925	335.475
6	240.875	241·450	235·100	240.525
5	270.025	283.850	270·175	267.325

H'quick

If FMT already loaded, start with **MWOD & OPDAY Load** drill (N-37):

Intentionally Blank

Aug 12

MWOD & OPDAY Load

Manual/Preset PRESET

Channel Selector Chan 20 (Beep)

Frequency Selectors 220.025

Preset Button (Red) Press and release (Beep)

Manual/Preset MANUAL Channel Selector Chan 20

Frequency Selectors Set first MWOD segment Tone Button Press and release (Beep)

Repeat last 3 steps with decreasing channel numbers until:

Channel Selector Chan 14

Frequency Selectors Date tag 3dd.000 (dd = today's

date)

Tone Button Press and release (Double

Beep)

Channel Selector Chan 1

Tone Button Press and release (Beep)

Manual/Preset PRESET

Channel Selector Chan 20 (Beep)

Frequency Selectors 220.000

Preset Button (Red) Press and release (Beep)

TRAINING WODS

Odd Nymharad daya Eyan Nymharad daya						
Odd Nur	Odd Numbered days		Even Numbered days			
SEG	Frequency	SEG	Frequency			
1	300.050	1	300.025			
2	386.600	2	339-425			
3	280.375	3	386.600			
4	339.425	4	280.375			
5	309.500	5	263.325			
6	263·325	6	309.500			
Date Tag	01	Date Tag	02			

TOD Acquisition

Frequency Selectors Set TOD distribution frequency 100 MHz Selector. Select T then release

TOD transmission. Tone indicates reception

Arbitrary Clock Start

100 MHz Selector.......... Select T and hold Tone button Press and release

100 MHz Selector. Release

TOD Send

Frequency/Net number. Set as required Tone Button Press, tone indicates transmission

Fault Tones

Constant Tone MWOD or TOD absent Pulsed Tone Invalid net number

Net Number Selection

Manual/Preset MANUAL

Frequency Selectors Set A+ net no (eg A00.925)

Return to Operate Drill

If radio becomes unresponsive, carry out **Return to Operate Drill**.

Manual/Preset PRESET Channel Selector Chan 20 Frequency Selectors 220.000

Preset Button (Red) Press and release (Beep)

A/F, T/R

AIRCREW LANDAWAY FLIGHT SERVICING SCHEDULE

IMPORTANT

Hazard and maintenance information is to be complied with throughout the work detailed in this schedule. If any problems or faults are encountered while under-taking any servicing, advice/assistance is to be sought from a qualified Hawk tradesman.

WARNING

That which if not observed may result in loss of aircraft and/or death or injury

CAUTION

That which if not observed may result in damage to aircraft or its equipment

NOTE

That which is essential to emphasize

COCKPIT

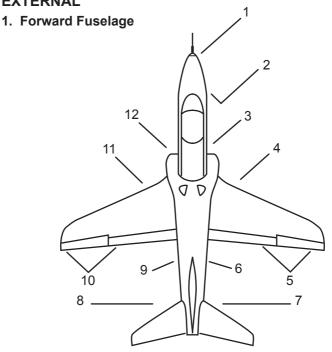
Both Cockpits (B), Front (F), Rear (R) (B) Oxygen regulators 100%, Off

	(B) Alt, Al &HSI Velcro secure and	
	stowed	All
		AII
	(R) WMP NORMAL/OVERRIDE	∧ /⊏ ⊤ /□
	switchNORMAL	A/F, T/R
	(B) Stby Inst NORMAL/BATT	^ /C T/D
	switch NORMAL	A/F, T/R
	(B) Accelerometer Check. Ensure no	
	overstress (+9/-4·5g)	A /F T/D
	Reset	A/F, T/R
	(B) Oxygen shut-off valve Set OFF	A/F, T/R
Servicing	(R) Canopy torque shaft	
	mounting bracket Check for cracks	All
	(B) Cockpit floor Clean and dry	All
	(B) ISIS combining glass Ensure clean and	
	no damage	All
	(B) Sight head roll test covers . Ensure secure	A/F, T/R
	(F) Batteries Check voltage (25V min)	B/F
	(F) Target towing release unit Operate (if fitted)	B/F, T/R
	(B) Gunsight heads No discolouration of	
	desiccant	B/F
	(B) Emergency internal lighting Check operation if	
	night flying	B/F, T/R
	(F) MASS Ensure SAFE.	
	Remove and stow key	A/F, T/R

F, T/R F, T/R F, T/R F, T/R F, T/R F, T/R F, T/R F, T/R
F, T/R F, T/R F, T/R F, T/R F, T/R F, T/R
F, T/R F, T/R F, T/R F, T/R F, T/R
F, T/R F, T/R F, T/R F, T/R F, T/R
F, T/R F, T/R F, T/R F, T/R
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A/F, T/R
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EXTERNAL



Serv cont..

Figure 1: Walk Round Route

NOTE: Aircraft external areas include all probes, aerials, intakes, exhaust ducts, pylons and pod tank (if fitted)

Cockpit drain valves Depress plunger to drain	A/F
Low cycle fatigue	
counter (if fitted) Record readings	A/F
Landing lamp window Look for damage	A/F, T/R
Fatigue meter Record readings	All
Pitot probe Look for damage,	
blockage and alignme	ent All

EXTERNAL - continued

•			
~	NOSA	Underca	rriand
4.	11036	Ollucica	nnage

Nose U/C doors Look for damage Nose U/C door mechanism Check secure & mechan	A/F, T/R nically
locked Nose U/C compartment Look for damage	A/F, T/R
and leaks	All
Nose U/C unit Look for damage and leaks	All
Shock Absorber Extension normal.	7 (11
Black line on fork below red sector	
on pivot bracket	All
Nosewheel assembly Look for damage Check pressure	
8.6 bar (+0.7/ -0 bar)	B/F
Check witness marks align	A/F, T/R
Oxygen system Replenish A/R	All
3. Starboard Forward Fuselage	
Engine air intake and	
LP compressor Look for damage	All
4. Starboard Main Landing Gear Bay	
Main undercarriage unit Look for damage	A II
and leaks Shock absorber Extension normal	All All
Clean inner strut	A/F
Mainwheel assembly Look for damage & wear. Check	
ծ wear. Cneck hub alignment marks	All
Check tyre pressure	
10.5 (+0.7/-0) bar Brake unit Look for damage and	B/F
leaks. Check pad	
wear, brakes ON	A/F, T/R
Leg fairing doors Look for damage Uplock spigot Clean	A/F, T/R B/F, T/R
Wheel brake accumulator Check 82 to 90 bar	B/F, T/R
Hyd 1 accumulator Check 72 to 80 bar Undercarriage	B/F, T/R
Compartment Look for damage & leaks	All

A/F, T/R

EXTERNAL - continued

4. Starboard Main Landing Gear Bay- continued	
Main U/C retraction jack indicator pin Ensure protruding Inner door Look for damage Inertia switch Ensure not operated Fire extinguisher Check pin not protrudir	A/F, T/R A/F, T/R A/F, T/R ag All
5. Starboard Wing CBLS no.100 Mk 3 Look for damage LAU-7A launcher Look for damage Launcher adaptor Look for damage AIM-9L Acquisition Round Look for damage Nitrogen receiver Check pressure Flap hinge fairings Look for cracks Flap shroud panels Check for delamination	A/F, T/R A/F, T/R A/F, T/R AII AII A/F, T/R A/F, T/R
6. Starboard Centre Fuselage	
Hyd 1 reservoir	B/F
4.5 to 4.6L	A/F, B/F
Hyd 1 flying control filter indicator	A/F, T/R A/F, T/R A/F, T/R A/F, T/R A/F A/F, B/F A/F, T/R A/F, T/R
7. Starboard Rear Fuselage	
Lower strobe light Free from contamination and moisture Tailplane deflector plates Look for damage Jet pipe	e, turbine

Airbrake Look for damage

Serv cont...

EXTERNAL - continued

7. Starboard Rear Fuselage- continued	
Oil tank Ensure level between '2 Pints' and Full.	
Replen if required	A/F, T/R
8. Port Rear Fuselage	
Towed target release unit If fitted, check for damage	A/F, T/R
Tailplane deflector plates Look for damage	A/F, T/R
9. Port Centre Fuselage	
Hyd 2 reservoir	B/F A/F, B/F A/F, B/F
Hyd 2 flying control filter	A/1 , D/1
indicator	A/F, T/R A/F, T/R
10. Port Wing	
CBLS no 100 MK 3 Check for damage LAU-7A launcher Look for damage Launcher adaptor Look for damage AIM-9L Acquisition Round Look for damage Nitrogen receiver Check pressure Flap shroud panels Check for delamination Flap hinge fairings Look for cracks	A/F, T/R A/F, T/R A/F, T/R AII AII A/F, T/R A/F, T/R
11. Port Main Landing Gear Bay	
Flap Emergency Cylinder Check 207 bar (+/- 3.4 bar)	B/F, T/R
U/C Emergency Cylinder Check 227 bar (+/- 3.4 bar)	B/F, T/R
Hyd 2 accumulator Check pressure Main undercarriage unit Look for damage	B/F, T/R
and leaks Shock absorber Extension normal Clean inner strut	All All A/F
Inertia switch Ensure not operated Throttle teleflex control	A/F, T/R
conduitLook for damage	A/F, T/R

Iss 6 AL 1 Mar 13

EXTERNAL - continued

Port Main Landing Gear Bay - continued	
Undercarriage	
Compartment Look for damage	
and leaks	All
Inner door Look for damage	A/F, T/R
Uplock spigot Clean	B/F, T/R
Leg fairing doors Look for damage	A/F, T/R
Brake unit Look for damage and	
leaks. Check pad	
wear, brakes ON	A/F, T/R
Mainwheel assembly Look for damage &	
wear. Check wheel	
hub alignment	Λ ΙΙ
witness marks.	All
	B/F
10.5 (+0.7/-0) bar	D/F
12. Port Forward Fuselage	
Engine air intake and	

LP compressor Look for damage All

Serv cont..

ANNEXES

Annex A - Pressure Refuelling

WARNING: Aircraft are not to be refuelled within 30mins

of GTS Start

Battery master switch.....ON

Fuel Note contents

Battery master switch..... OFF Front cockpit throttle..... HP OFF

Refueller Ensure earthed, bond to aircraft.

Note contents and set delivery

meter to zero.

Refuelling panel Open access door and close

fasteners on panel to prevent damage from refuelling hose. Connect refuelling hose to

aircraft.

Refuelling switch REFUEL

CAUTION: Refuelling is to be stopped if excessive fuel discharges through either the relief valve on the port fuselage side or the fuel vent on the tail cone. Refuelling pressure is not to exceed 3.45 bar(50psi). Refuelling flow rate is not to exceed 680 l/min (150gal/min).

Refueller...... Start delivery
Pre Mod 2010 ensure air vents from relief valve
Post Mod 2010 ensure NO air vents from relief valve
Ensure air vents from tail cone vent

When complete

Refueller..... Stop delivery. Ensure Refuelling

Indicator Lamp extinguished.

Refuelling switch OFF

Refuelling panel Disconnect refuelling hose from

aircraft. Ensure cap correctly fitted/locked. Close panel

Refueller Disconnect bonding lead. Note

contents and delivery meter

reading.

Battery master switch..... ON

Fuel Note contents

Battery master switch..... OFF

Annex B - Hyd 1 & 2 Fluid Contents and Nitrogen Pressure Check

Ensure U/C & flaps down and airbrake in Dissipate hydraulic pressure by operating control column Release brake pressures

HYI	HYDRAULIC RESERVOIR INFLATION AND CHARGING DATA					
	-26°C to 0°C		0°C to 30°C		+30°C to +60°C	
	Pressure Bar +/- 0.1 PSI +/- 1.5	Volume +/- 0.1L	Pressure Bar +/- 0.1 PSI +/- 1.5	Volume +/- 0.1L	Pressure Bar +/- 0.1 PSI +/- 1.5	Volume +/- 0.1L
Hyd 2	2.2 32	4.1	2.7 39	4.2	3.2 48	4.3
Hyd 1	2.7 39	4.3	3.3 48	4.6	4.0 58	4.8
Wheel-brakes accumulator charged to 207 bar (see below)						
Hyd 1	2.2 32	3.5	2.7 39	3.7	3.3 48	4.0

Annex C - 'Popped' Hyd Filter Procedure

If the Hyd 1 Front/Rear or Hyd 2 Front/Rear filter indicators are 'popped', proceed as follows:

Check F700 Section 1 SNOWS

If the filter has 'popped' within 5 hrs, seek engineering advice. If this is the first pop or it is greater than 5 hrs from the previous pop, reset the indicator, raise a SNOW for the pop.

Example - "Hyd 1 Rear filter indicator popped - RESET" Record flying hours and Aircrew Accept. If in doubt seek engineering advice

Annex D - ADR Status Check

Ignition switch.....NORMAL

Serv cont..

Annex E - Aircraft Minimum States

SYSTEM	B/F	T/R	A/F
GTS	No requirement for aircrew to replenish		
ENGINE OIL	N/A	4.5	6.5
HYD 1 SYSTEM	4.5 (+/- 0.1L)	4.5	4.6*
HYD 2 SYSTEM	4.2 (+/- 0.1L)	4.1	4.2
OXYGEN#	6+	4+	7+

^{*} Ensure brake pressure released, flaps down and airbrake in # Maximum Oxygen regulator delivery pressure 1800 psi

Annex F - Additional Information

To pump up brake pressure use hand pump located behind access panel on the starboard side of the air intake. If the airbrake is down it will retract as soon as hand pumping of the brake system starts.

Annex G - Parking Overnight

Engine air intake blanks Fit on A/F, remove on B/F
Jet pipe blanks Fit on A/F, remove on B/F
U/C Ground Locks Fit on A/F, remove on B/F
Weapon Pins Fit on A/F, remove on B/F
Note: With the parking brake left on system pressure will dissipate

Note: With the parking brake left on system pressure will dissipate with time, therefore ensure the aircraft is chocked.

Annex H - RAFAT Aircraft

Carry out the following additional checks to RAFAT aircraft		
Smoke master switch OFF	All	
Smoke selected indicator Out	All	
Tailcone assembly Check for damage		
and leaks	All	
Engine Life Computer Carry out Self		
Test (AMM)	All	

Annex J - Completing F705 (Hawk)

- 1. Ensure TDM for previous AF and Sheet No recorded
- 2. Line 6 Enter servicing type and strike through 'T'
- 3. Line 7 Enter TDM for start time
- 4. Line 8 & 9 Sign both Man A & Man B blocks
- 5. Lines 15 21 Sign for spare slots as required
- 6. Line 23 For BF and TR enter validity of servicing
- 7. Line 25 Enter last SNOW from F707A

7.	7. Line 25 Enter last SNOW from F707A					
				(a))	
			Na	ame	Sig]
1		Accepted Faults			1	Score Z through boxes 1 to 5
2	After Flight	Signature	J BI	oggs	J Bloggs	in column if B/F
3	Declaration	TDM		1015/1	8/08	Local Time
4		Smoke Used		W B	R]
5	Armament Clea	arance				
6	Flight Servicing	g	B/F / 1	TR / AF	T/A	A/F, B/F, T/R as reqd
7	Commenced T	DM		1030 / 1	8 / 08	Time started
8	ı	Man A	J BI	oggs	J Bloggs	Sign for
9	N	Man B	J BI	oggs	J Bloggs	Man A & Man B
10	A	vionics				Score Z through
11	El	ectrical				boxes 10 to 12 in column for
12	W	eapons				servicing
13	Oxygen		/8			Oxy contents in 1/8s
14	4 Pannier		F/R			Signature lines 13 to 21
15	15 1. Spare					
16	16 2. Spare					
17	17 3. Spare					
18	18 4. Spare					
19	5. Spare					
20	6. Spare					
21	7. Spare					
22	Flight Servi	cing Co-odinator	J BI	oggs	J Bloggs	A/F 72 hrs,
23	Valid	until TDM		2230 / 1	8 / 08	B/F & T/R 12 hrs
24	Fina	al Arming]
25	Las	st SNOW]
26	MOD Form 700C Co-ordinator		J Bloggs J Bloggs		J Bloggs	
27	27 Co-ordinated TDM		1045 / 18 / 08		8 / 08	Score Z Through
28	Flying F	Requirement				boxes 24 to 31 in column if A/F
29	Aircrew	Aircrew Accepted Faults				
30	Acceptance Certfifcate	Signature	J BI	oggs	J Bloggs]
31		TDM		1230 / 1	8 / 08	<u> </u>

Serv cont..

Situation	Servicing	Validity
After landing - Next Flight within 12 hrs	T/R	12 hrs
After Landing - next Flight outside 12 hrs	A/F	72 hrs
Within 12 hrs of next flight	B/F	12 hrs

The aircraft must have a valid A/F and B/F or a T/R to fly. The A/F is invalid on landing unless a T/R is carried out. The A/F is invalid if the aircraft does not fly before the expiry of a T/R.

F705 (Hawk) reverse - Fuel Certificate

Record fuel state as per example

1	Refuel/Det	fuel/Check	(Delete inapplicable activity
	Fuel Remaining	750		Fuel remaining on A/C gauges
	Total put in/ Taken out	600kg		
	Total in A/C	1350kg		
1	etered Put in/ ource Taken Out* 580		Bowser Reading	
Dis	Discrepancy 20kg 3%		Limits 3% or 40kg	
Туре:				
Signature				
	TDM 10451808			

Strike through and sign for apron F/R

F705(SSC) - Supplementary Servicing Certificate

Ensure that the carried forward details are entered correctly. Sign for each check detailed on the F705(SSR)

F726HSSR(Hawk) - Hydraulic System Supplementary Record Record any hydraulic accumulator 'top-ups'.

F725C - Flying Log and Fatigue Data Sheet

Ensure A/C no, Sheet No and Brought Forward Readings recorded from previous F725

Ensure sortie details are accurately recorded.

Ensure Fatigue Meter readings entered are sensible and count forward from previous readings.

F706B - Fuel Uplifts not Containing FSII or Lubricity Additive

Record any uplifts that do not contain FSII

E-2	EMERG GROUND EGRES	SS, RUNWAY	
E-3	SURGE, IMMEDIATE REL	LIGHT	T6NL
E-4	ASSISTED & UNASSISTE	ED RELIGHT	
E-5	ENGINE FIRE		FIRE
E-6	AIR PROD FIRE, GND EM	ERG S'DOWN	START
E-7	ENGINE FAILURE	T6NL ECA	E OHT
E-8	SMOKE OR FUMES, OXY	GEN .	CPR
E-9	DIFFICULTY BREATHING	3	OXY
E-10	GENERATOR		GEN
E-11	HYDRAULICS	HYD HYD 1	HYD 2
E-12	HYDRAULICS	HYD 1	HYD 2
E-13	JETPIPE, OIL	JPOHT	OIL
E-14	FUEL	F.PR FUEL	TRANS
E-15	INVERTER	AC 1 AC 2	AC 3
E-16	ANTI-SKID, RUNWAY		SKID
E-17	FORCED LANDING		
E-18	EJECTION		
E-19	MDC, JETTISONING		
E-20	LANDING GEAR		
E-21	LANDING GEAR		
E-22	LANDING WITH GEAR U	NSAFE	
E-23	FLAPS, AIRBRAKE		
	TRIM		
E-24			
E-25	COMMUNICATIONS LOW SPEED HANDLING	CHECK	

EMERGENCY GROUND EGRESS

WARNING 1: If the decision to eject is made, canopy must be closed and locked

WARNING 2: The manual separation handle must not be used for ground egressing

If necessary

1. Eject

O	th	er	W	S	e:

- 1. Throttle HP Off
- 2. Seat firing handle

safety pin Insert correctly

- 3. Canopy Open/Operate MDC (E-19)4. Unstrap completely:
- - a. QRF Release
 - b. PEC & PSP Disconnect
 - c. Leg restraints......Pull through/release garters
- 5. Aircraft Vacate & move clear upwind

ABORT

- 1. Throttle Idle
- 2. Brakes Apply
- 3. Flaps Down
- 4. Control column. Fully back below 100 knots
- 5. Throttle Consider HP Off

BARRIER ENGAGEMENT

- 1. Throttle HP Off
- 2. Head...... Duck before engagement
- 3. Brakes Release before entering barrier

When aircraft stopped:

4. Continue with **GROUND EMERGENCY ENGINE SHUTDOWN**

OVERRUN / RUNWAY DEPARTURE

Consider the following options:

1. Eject

Or:

1. Throttle HP Off

When aircraft stopped.

2. Continue with **GROUND EMERGENCY ENGINE SHUTDOWN**

Or:

1. Landing Gear Emergency Retraction

Note 1: If emergency (maximum) braking has been used, the aircraft should not be taxied

Note 2: The aircraft is cleared to enter Mk 6 and Type A barriers, and cleared to enter Mk 12a and Type B barriers set to the light aircraft setting

AP 101B-4401-14

Issue 6 Aug 12

ENGINE SURGE with or without T6NL

Rapid increase in TGT and/or unusual engine noise, possible loss of thrust and/or stagnating RPM

Abort Surge Relight E-3 1. Throttle Idle 2. TGT Monitor

Note: TGT may take up to 6 seconds to stabilise

If idle TGT > 450°C:

3. Carry out IMMEDIATE RELIGHT (below)

If idle TGT < 450°C:

If T6NL illuminated during surge or (685°C exceeded):

- 3. Prove engine to minimum RPM required
- 4. Land ASAP

If T6NL did not illuminate during surge (685°C not exceeded)

3. Prove engine and use normally

IMMEDIATE RELIGHT

- 1. Throttle HP Off
- 2. Relight button..... Press and hold
- 3. Throttle Idle (against idle stop)
- 4. Relight button..... Release
- 5. Rotation MI If not Green, press and hold Relight button
- 6. Airbrake......In

If no relight within 30 seconds or if 585°C TGT is likely to be exceeded (Note 1):

7. Throttle HP Off

Note 1: Under transient conditions 585°C may be exceeded by up to 20°C for 10 seconds

Note 2: If practicable, wait 30 seconds for engine to drain before attempting another relight *If engine relights normally:*

7. Carry out **POST RELIGHT** (below)

POST RELIGHT

- 1. Throttle Prove engine surge free to minimum RPM required
- 2. Electrics Reset
- 3. Hyd 2 Reset button. Reset with RPM above 76%
- 4. Eng Start Off for 5 secs min, then On
- 5. GTS MI..... Confirm Black

If the T6NL illuminated or 685°C was exceeded or engine damage is suspected:

6. Land ASAP

If the T6NL did not illuminate or 685°C was not exceeded:

6. Land As soon as practicable

ASS	IST	FD	RFI	IGHT
AJJ		$ \nu$	Γ	.iGiii

1. Speed. 165 to 250 knots

If GTS Black:

2. Altitude Below FL 200

5. Ignition Normal

6. Relight button..... Press and release

7. Continue with GTS Green (below)

If GTS Green:

2. RPM..... Below 20%

3. Eng Start Start and release

5. Rotation MI. If not Green, press and hold Relight button

If no relight within 45 seconds from selecting Start or if 585°C TGT is likely to be exceeded (Note 1):

Note 1: Under transient conditions 585°C may be exceeded by up to 20°C for 10 seconds

Note 2: If practicable, wait 30 seconds for engine to drain before attempting another relight

If engine relights normally:

6. Carry out **POST RELIGHT** (E-3)

UNASSISTED RELIGHT

1. Altitude Below FL 250

5. Eng Start switches Both ON6. Ignition Normal

8. Throttle..... Idle (against idle stop)

9. Relight button..... Release

10. Rotation MI. If not Green, press and hold

Relight button

If no relight within 30 seconds or if 585°C TGT is likely to be exceeded (Note 1):

11. Throttle..... HP Off

Note 1: Under transient conditions 585°C may be exceeded by up to 20°C for 10 seconds

Note 2: If practicable, wait 30 seconds for engine to drain before attempting another relight

If engine relights normally:

11. Carry out POST RELIGHT (E-3)

ENGINE BAY FIRE IN FLIGHT



- 1. Throttle Minimum practicable / Idle
- 2. Check for further signs of fire

If fire is confirmed:

Relight Fire E-5 3. Throttle HP Off

4. Fuel Pump...... Off 5. LP Cock..... Off

6. Speed Reduce to minimum

practicable (ideal < 250kts)

7. Extinguisher button.... Press

8. Oxygen 100%

9. Mask Toggle down

10. Cabin conditioning Off

If FIRE remains on after 45 seconds or if evidence of fire:

11. Eject

If FIRE goes out:

11. Consider **FORCED LANDING** (E-17)

or **PREMEDITATED EJECTION** (E-18)

If fire is not confirmed (FIRE on or off):

4. If use of power illuminates **FIRE**, check frequently that caption goes out on throttling back

ENGINE BAY FIRE ON THE GROUND



1. Throttle HP Off

2. Fuel Pump..... Off

3. LP Cock Off

4. Extinguisher button.... Press

5. Batteries Off

6. Carry out EMERGENCY GROUND EGRESS (E-2)

AIR PRODUCER BAY FIRE - IN FLIGHT

START

- 1. Eng Start.....Off
- 3. Height Above 5000ft AGL
- 4. Check for confirmatory signs of fire
- 5. Flap Select Mid if required
- 6. Monitor tailplane control

If tailplane control is lost:

- 7. Eject
- If START remains on:
- 7. Land after 20 minutes
- If START goes out:
 - 7. Land after 10 minutes

AIR PRODUCER BAY FIRE ON THE GROUND

START

- 1. Eng Start.....Off
- 2. Throttle HP Off
- 3. Fuel Pump..... Off
- 4. LP Cock..... Off
- 5. Batteries Off
- 6. Carry out EMERGENCY GROUND EGRESS (E-2)

GROUND EMERGENCY ENGINE SHUTDOWN

Any of the following:

T6NL E OHT HYD JPOHT OIL ECA HYD 1 HYD 2

or Symptoms of Mechanical failure

- 1. Throttle HP Off
- 2. Fuel pump switch..... Off
- 3. LP Cock..... Off
- 4. Batteries Off

Note 1: Hyd captions may be associated with a fire risk

Note 2: If signs of fire carry out EMERG GROUND EGRESS (E-2)

Oct 17

ENGINE SEIZURE / MECHANICAL FAILURE

WARNING: Do not attempt to relight a seized engine

- 1. Throttle HP Off
- 2. Fuel Pump..... Off
- 3. LP Cock..... Off
- 4. Speed...... Below 300kts/0.6 Mach (If no windmilling Hyd 1 pressure)
- Consider FORCED LANDING (E-17) or PREMEDITATED EJECTION (E-18)

Start Engine E-7

ENGINE LP COOLING AIR OVERHEAT

E OHT (Pre MOD 2010 aircraft)

In flight:

- 1. Throttle Minimum practicable
- 2. Land ASAP

Note: Post MOD 2010 aircraft, spurious warning - Ignore

T6NL OVERLIMIT/CONTROL AMPLIFIER FAILURE

T6NL and/or ECA

- 1. Throttle Set approx 80% RPM
- 2. TGT Monitor

If TGT does not respond normally (approx 400°C) and/or T6NL caption remains illuminated:

- 3. Throttle Idle
- 4. TGT Monitor

If TGT more than 450°C:

5. Carry out **IMMEDIATE RELIGHT** (E-3)

If TGT less than 450°C:

- 5. Throttle Prove engine to minimum RPM required (see WARNING)
- 6. Land ASAP

If TGT responds normally (approx 400°C):

- 3. Throttle Prove engine to minimum RPM required (see WARNING)
- 4. Land As soon as practicable

WARNING 1: Use gentle throttle movements

WARNING 2: Max 95% RPM (90% above 20,000ft)

WARNING 3: Min 70% RPM until committed to land if ECA caption remains illuminated

SMOKE OR FUMES

2. Ma 3. De	kygen	Toggle down
 Alt Ca Ca 	titude	Off Below 25,000ft
<i>If smo</i> 8. Sp	ke or fumes remain and p	oresent a danger Reduce to minimum practicable

SUSPECTED HYPOXIA AND/OR CONTAMINATED OXYGEN SUPPLY

1. Connections and mask seal......Check

If symptoms persist or oxygen supply is not restored

- 2. Emergency oxygen ring . Pull
- 3. Main oxygen supply.... Off
- 4. Cabin altitude Below 10,000ft
- 5. Land As soon as practicable

Note 1: When emergency oxygen is operated, no cockpit indications show the system is operating or the contents remaining

Note 2: Disconnect main oxygen hose when supply is exhausted

Note 3: If PEC is adjusted, ensure leg restraints are reconnected

CPR CABIN ALTITUDE ABOVE 25,000FT

- 1. Mask Toggle down
- 2. Cabin conditioning Flood
- 3. Cabin altitude Below 25,000ft

OXY or MI CONTINUOUS WHITE OR BLACK or DIFFICULTY BREATHING IN

1. Connections and mask seal Check

- 2. Oxygen selector Confirm On
- 3. Oxygen contents Check

If oxygen contents are above zero:

4. Oxygen regulator 100%

If oxygen contents are zero, breathing is restricted or supply is not restored:

- 5. Emergency oxygen ring . Pull (if required)
- 6. Cabin altitude Below 10,000ft
- 7. Land As soon as practicable

When emergency oxygen is exhausted:

8. Main oxygen hose Disconnect

If breathing normal:

- 5. Oxygen contents and MI. Monitor
- 6. Continue sortie

DIFFICULTY BREATHING OUT

- 1. Mask.....Lift off face
- 2. Oxy Test button Press

If difficulty persists:

- Breathe in through mask and exhale with mask away from face
- 4. Cabin altitude Below 10,000ft

Fumes CPR

Oxygen

E-9

GENERATOR FAILURE

GEN Voltmeter approx 24V
F.PR AC 1
AC 2
AC 3 **

GEN Voltmeter 27 to 29V

- 1. Spurious warning
- 2. Monitor voltmeter

DC Reset button. Press. If successful go to
 Successful Reset

2. Negative g Avoid

3. Throttle Avoid high power settings

If no reset:

4. Stby Inst switch Batt

5. UHF power switch Batt

6. In turn, switch Batt 1 & 2 Off then On. If voltage increases with one switch Off, leave that switch Off

7. DC Reset button. Press. If successful go to Successful Reset

If no reset:

8. Electrics Load shed (below); monitor voltage

9. Consider descent to VMC below cloud

Before voltage drops to 21V:

If range acceptable with landing gear and flaps lowered:

10. Landing gear and Flaps . Lower

If range critical with landing gear and flaps lowered:

10. Cockpit Emergy Lights . . As required

11. Batteries Off until required

Successful Reset

1. AC Reset if required

If F.PR illuminated after 10 seconds:

2. Fuel Pump..... Off then On

LOAD SHEDDING

Consider turning off:

Pitot heater Landing lamp

External lights

Internal lights

Out of use radio IFF/SSR

GVRS

The stronger battery

CAUTION: With both Batt 1 & 2 off, all essential electrical services lost, including:

Relighting VHF radio
ECA monitoring Standby UHF

Pitot heater IFF/SSR

CWP indications Flap & L/G normal selection

Anti-skid & indication

Trims Airbrake selection

Iss 6 AL3

TOTAL HYDRAULIC FAILURE

HYD 1

HYD 2

- Both Hyd gauges below 45 bar

1. Eject

HYD 1 FAILURE

HYD 1 - Hyd 1 gauge below 45 bar

- 2. Hyd pressures Monitor
- 3. Airbrake Do not use
- 4. Speed..... Avoid high speeds
- 5. Landing gear & Flaps Select normal DOWN then lower on standby system
- 6. Land ASAP

Generator Hydraulics E-11 8. Brakes Single gentle application, avoiding anti-skid operation

HYD 2 FAILURE

HYD 2 - Hyd 2 gauge below 113 bar or cycling between 160 to 210 bar

- 1. Airbrake......In
- 2. Hyd pressures Monitor

If RAT is not operating (Hyd 2 pressure below 103 bar):

- 3. Airbrake Do not use
- 4. Speed..... Avoid high speeds
- 5. Land ASAP

If RAT operating (Hyd 2 pressure cycling between 160-210 bar) and HYD 2 is illuminated at all power settings:

- 3. Airbrake Use only if necessary
- 4. HYD 2 pressure Monitor
- 5. Land ASAP

If RAT operating (Hyd 2 pressure cycling between 160-210 bar) and HYD 2 only illuminates below 76% (uncommanded RAT):

- 3. Hyd 2 Reset Press to reset RAT
- 4. Hyd 2 pressure Check correct operation
- 5. Continue sortie

HYD 1 AND HYD 2 FAILURE

HYD 1 - Hyd 1 pressure below 45 bar

HYD 2 - Hyd 2 pressure cycling between 160 to 210 bar

- 1. Airbrake..... In
- 2. Hyd pressures Monitor
- 3. Airbrake Do not use
- 4. Speed..... Below 300kts/0.6M
- 5. Landing gear & Flaps Select normal DOWN then lower on standby system
- 6. Land ASAP, threshold speed 125 kts

After landing:

- 8. Brakes Single gentle application, avoiding anti-skid operation

HYD 1 WHEN LANDING GEAR SELECTED UP

- 1. Landing gear..... Down
- 2. Speed..... Below 200kts

If HYD 1 remains illuminated:

- 3. Carry out **HYD 1 FAILURE** (E-11)

Note 1: Selecting landing gear DOWN immediately may prevent loss of fluid from the system

Note 2: Do not reselect landing gear UP

UNDEMANDED FLAP LOWERING

Flaps lower without being selected from either cockpit

- 1. Flap Down
- 2. Speed..... Below 200kts

Note 1: Do not select flap UP

Note 2: HYD 1 may illuminate. Selecting flap down immediately reduces possible loss of hydraulic fluid and may cause HYD 1 to go out

AP 101B-4401-14 Issue 6

Issue 6 Aug 12

JET PIPE BAY OVERHEAT

JPOHT

- 1. Throttle Minimum practicable
- 2. Land ASAP

OIL PRESSURE LOW

OIL

1. Throttle	. Smoothly set between 78 - 87%
2. RPM	. Monitor and maintain with
	smooth, progressive throttle
	movements
3. Maintain 1g nominal	
4	A C A D

WARNING 1: Loss of oil pressure will result eventually in engine seizure

WARNING 2: Bearing loads increase as altitude reduces

WARNING 3: Increasing vibration is indicative of early failure

WARNING 4: The optimum RPM range to prolong engine life is

80-85%

Hydraulics JPOHT Oil E-13

E-13

LOW FUEL PRESSURE

F.PR

WARNING: There may be a fuel leak and consequently a fire risk

1. Throttle Minimum practicable

2. Negative g..... Avoid

3. Fuel Pump Confirm On

4. Fuel contents Monitor

5. Land ASAP

LOW FUEL CONTENTS

FUEL

WARNING: If more than 160kg indicated, assume only 160kg useable

1. Throttle Minimum practicable

2. Fuel contents Check

If more than 160kg indicated and F.PR and TRANS captions out:

3. Apply negative g then positive g

If warning persists:

4. Land ASAP

LOW FUEL TRANSFER PRESSURE

TRANS

1. Altitude Below 25,000ft

2. Negative g Avoid

3. Throttle Avoid high power settings

4. Land As soon as practicable

Note 1: Maximum power may be limited at high altitude

Note 2: FUEL may illuminate with fuel contents indicating > 160kg

HIGH FUEL CONSUMPTION

WARNING: There may be a fuel leak and consequently a fire risk *If fuel discharge from main vent can be confirmed:*

1. Apply negative then positive g

If venting continues:

2. Land ASAP

If venting stops:

2. Land As soon as practicable

If fuel discharge from main vent cannot be confirmed:

1. Land ASAP

Iss 6 AL3 Oct 17

DOUBLE INVERTER FAILURE

AC 1 & AC 2

1. AC 1 Reset button Press

If no AC 1 reset:

2. AC 2 Reset button Press

If either reset successful:

3. Land As soon as practicable

Note: Do not attempt further resets

If AC 1 & AC 2 remain illuminated:

3. Repeat resetting procedure after a few minutes (max 3 resets)

4. Land As soon as practicable

CAUTION: If both inverters remain off line the following flight instruments are unusable:

Main Al

HSI

AHRS

Main Altimeters

Fuel Inverters E-15

SINGLE INVERTER FAILURE

AC 1 or AC 2 or ** AC 3

1. Appropriate Reset button . . Press

If unsuccessful:

2. Repeat resetting procedure after a few minutes (max 3 resets)

SKID ANTI-SKID FAILURE or LOSS OF BRAKING / UNEXPLAINED VIBRATION / DIRECTIONAL SWING AT ANYTIME

- 1. Brakes Release
- 2. Anti-Skid..... Off
- 3. Brakes Reapply

If dual:

4. Other cockpit brakes Consider use

If unsuccessful:

5. Parking brake Gentle application

WARNING: With Anti-Skid On, a tyre burst on landing can cause rapid and uncontrollable directional change

Note: If foot motor failure is suspected, gentle application of the parking brake may offer symmetric braking but without anti-skid protection

RESIDUAL BRAKE PRESSURE > 10 BAR

WARNING: With Anti-Skid On, a tyre burst on landing can cause rapid and uncontrollable directional change

If unable to ascertain aircraft mod state, action the pre-mod drill.

Pre-mod 1863 aircraft:

1. Anti-Skid..... Off

Note 1: With Anti-Skid Off, a tyre burst can cause a directional change which is controllable with opposite brake

Post-mod 1863 aircraft:

1. Anti-Skid..... Leave On

Note 2: Anticipate a swing towards the high brake pressure after the touchdown protection system allows the brake pressure to rise

FORCED LANDING

-		
	470/475	80 knots clean (0.65M above FL200) 165/170 knots gear down
ANA 1	• t r	(Use the higher speed if more than 1000kg fuel remaining or more than 500kg with external
		stores) 5000 kg AUM)
-	2. Airbrake Ir	· ·
	3. External stores J	` ,
	4. Hyd 1 & 2 pressures N	
	5. UHF power switch B	
	6. Stby Inst switch B	
	7. IFF/SSR	·
•	8. Electrics L	load shed as appropriate (E-10)
P	Approach and Landing	
	1. ThrottleH	HP Off
:	2. Fuel Pump C	Off
;	3. LP Cock C	Off
4	4. Eng Start C	Off
V	When appropriate:	
	5. Landing gear S I	Select normal DOWN then lower on standby system
(6. Harness L	ocked and tight
V	When appropriate:	
	7. Flaps	Select normal DOWN then
Forced-		lower on standby system
Landing E-17	8. Approach speed1	

PREMEDITATED EJECTION

WARNING: To avoid possible seat collision, simultaneous or near simultaneous ejections should be avoided

If time and conditions permit:

II LI	ine and conditions perinit.	
1.	Consider	Area of parachute landing and
		area of aircraft impact
2.	Other occupant	Warn
3.	Height	Ideally between 2000ft and
	9	9000ft AGL
4	Airspeed	Ideally 250 knots
	Transponder	
6.	Harness	Locked and tight, leg restraints
		and PSP lowering line connected
7.	Oxygen mask	Tight
8.	Visors	Down
9.	Aircraft	Trim as required. Consider the
		use of aileron trim (max ½). Aim
		towards selected impact area
0	Dadia	
	Radio	
10.	Throttle	Idle
11.	Eyes	Tightly closed
	Ejection position	
		, 100 di 110

FAILURE TO EJECT

1.	Seat pin	Ensure removed
2.	Seat firing handle	Pull again

13. Seat firing handle Pull. Use command eject

system if practicable

If this fails:

4. 5. 6. 7. 8.	MDC	Disconnect Pull through/release garters Operate (E-19) Trim fully forward and invert
9.	Manual separation handle	Operate (with seat firing handle pulled fully upwards)

AUTOMATIC SEAT SEPARATION SYSTEM FAILS AFTER EJECTION

1. Seat firing handle..... Ensure fully pulled

2. Manual separation handle.....Operate

MDC OPERATION

WARNING: Do not enter barrier after MDC has been operated

- 1. Other occupant. Warn
- 2. Visors Down
- 3. Seating posture Erect
- 4. MDC firing handle Grasp
- 5. Eyes Closed tightly
- 6. MDC firing handle Pull

JETTISONING EXTERNAL STORES

WARNING 1: Jettison on or near the ground may result in ricochet damage

WARNING 2: Jettison above 400 knots may result in pylon damage

- 1. Landing gear and Flaps . . . UP
- 2. Aircraft Maintain straight and level
- 3. MASS..... Live, (busbars indicate Black)

CBLS

Ejection MDC

Jettisoning E-19 1. WCP or WMP Jettison button..... Press (maximum 3 seconds)

SIDEWINDER

- 1. Throttle Minimum practicable
- 2. MCP jettison button Press
- 3. After jettison of missiles carry out breakaway manoeuvre

Note: Acquisition Sidewinder missiles cannot be jettisoned

GEAR SELECTED UP BUT FAILS TO RETRACT

- 1. SpeedBelow 200 knots
- 2. Hyd 1 pressure.....Check

If Hyd 1 pressure low:

- 3. Landing gear.....Down
 - 4. Carry out **HYD 1 FAILURE** (E-11)

If Hyd 1 pressure normal:

With 3 greens showing:

3. Check selection in other cockpit

With one red indication and if visual inspection confirms gear is up:

- 3. Select DOWN, check 3 greens
- 4. Reselect UP, check gear indicates UP

In all other cases:

- 3. Select DOWN (ensure button fully in), check 3 greens
- 4. Make no further UP selection

GEAR UP SELECTOR CANNOT BE DEPRESSED

- 1. Speed Below 200 knots
- 2. Attempt further UP selection

If fault persists:

- 3. Select DOWN......Button fully in, 3 greens
- 4. Make no further UP selection

GEAR SELECTED DOWN BUT ONLY PARTIALLY EXTENDS

- 1. SpeedBelow 200 knots
- 2. Hyd 1 pressure.....Check

If Hyd 1 pressure low:

3. Carry out HYD 1 FAILURE (E-11)

If Hyd 1 pressure normal:

3. Obtain visual check if possible

With mainwheel red and 2 green indications:

- 4. Select UP
- 5. While applying prolonged negative g, select DOWN
- 6. After DOWN selection apply positive g

If unsuccessful:

7. Make further selections (allow 10 secs between selections)

If still unsuccessful:

8. See LANDING WITH GEAR UNSAFE (E-22)

With any other combination of indications:

- 4. Select UP
- 5. Make further selections varying g and speed

If unsuccessful:

6. See LANDING WITH GEAR UNSAFE (E-22)

AP 101B-4401-14 Issue 6 Aug 12

GEAR SELECTED DOWN BUT FAILS TO MOVE

Gear indicates 3 x UP

- 1. Speed Below 200 knots
- 2. Hyd 1 pressure Check

If Hyd 1 pressure low:

3. Carry out **HYD 1 FAILURE** (E-11)

If Hyd 1 pressure normal:

- 3. Check selection in rear cockpit
- 4. Make further selections (allow 10 seconds between selections)

If gear still fails to extend (even partially):

5. Carry out **GEAR STANDBY - LOWERING** (below)

GEAR STANDBY - LOWERING

WARNING: Standby lowering may cause a Hyd 1 failure

- 1. Airbrake Confirm In
- 2. Landing gear Select normal DOWN
- 3. U/C standby handle Press central button and pull handle

CAUTION 1: Gear cannot subsequently be retracted

CAUTION 2: Gear / Airbrake interconnect does not function

Gear

E-21

LANDING WITH GEAR UNSAFE

- 1. *Release explosive stores in safe area
- 2. *Retain CBLS
- 3. UHF power switch Batt
- 4. Harness Locked and tight
- 5. Arrester cable..... Removed if possible

GEAR POSITION	RECOMMENDED ACTIONS
Nosewheel UP/unsafe Both main wheels DOWN	Caution: If arrester cable in place, do not cross cable with nose lowered 1. Make normal powered approach After touchdown: 2. Throttle HP Off 3. Lower nose onto runway by 70kts 4. Do not use wheelbrakes until nose is on the ground 5. Use brakes to keep straight
Nose wheel DOWN One main wheel unsafe	Caution: If arrester cable in place, do not touch down before cable 1. Choose runway with maximum clear area in direction of anticipated swing 2. Anti-Skid Off 3. Make normal powered approach After touchdown: 4. Lower nosewheel 5. Throttle
All gear UP All gear All gear unsafe	Caution: If arrester cable in place, do not touch down before cable 1. Carry out low speed handling check (E-26) at safe height with Down flap selected 2. Make normal powered approach (with Mid flap if Down flap causes handling difficulties) After touchdown: 3. Throttle HP Off 4. Batteries Off
All other configurations	Attempt to obtain one of the above configurations. If this is not possible - Eject

FLAPS FAIL TO OPERATE

- 1. Speed Below 200 knots
- 2. Hyd 1 pressure Check

If Hyd 1 pressure low:

3. Carry out **HYD 1 FAILURE** (E-11)

If Hyd 1 pressure normal:

- 3. Check selection in rear cockpit
- 4. Make further selections from both cockpits
- If flaps still fail to operate, consider FLAPS STANDBY LOWERING (below)

FLAPS - STANDBY LOWERING

WARNING: Standby lowering may cause a Hyd 1 failure

- 1. Speed..... Below 200 knots
- 2. Flaps..... Select normal Down
- 3. Flaps standby handle Press central button and pull handle

Caution: Flaps cannot subsequently be raised

AIRBRAKE FAILS TO OPERATE

1. Hyd 1 pressure Check

If Hyd 1 pressure low:

2. Carry out **HYD 1 FAILURE** (E-11)

If Hyd 1 pressure normal:

2. Make further selections from both cockpits

If airbrake is stuck out:

- 3. Speed..... Below 200 knots
- 4. Landing gear DOWN then UP

AIRBRAKE / LANDING GEAR INTERCONNECT FAILURE

Airbrake known to be out with landing gear DOWN:

- 1. Speed Below 200 knots
- 2. Landing gear UP
- 3. Airbrake In, confirm MI Black
- 4. Landing gear DOWN

Gear Unsafe Landing Drill If Airbrake Remains Extended: Flaps

- 1. Make a shallower than normal approach to the runway
- 2. Do not flare
- 3. Lower the nosewheel immediately after touchdown

E-23

MAIN TAILPLANE TRIM FAILURE

Main tailplane trim inoperative or runs away

- 1. Standby tailplane trim cover Lift
- 2. Use standby tailplane trim

RUDDER OR AILERON TRIM FAILURE

Rudder or aileron trim inoperative or runs away With runaway rudder trim:

- 2. Flap Select Mid if required

 Note 1: With runaway aileron trim, no control authority is lost

Note 2: With runaway rudder trim, loads proportional to IAS

LOSS OF INTERCOM OR RADIO RECEPTION

1. Helmet and PEC connections.....Check

2. Aerial selection..... Check

Note: If PEC adjusted, reconnect leg restraints

If service not restored:

3. Amplifier selector FAIL

If UHF or intercom not restored:

- 4. Amplifier selector NORM
- 5. UHF power switch Batt

If unable to restore UHF:

- 6. UHF power switch Norm
- 7. Use alternative radio

If UHF, VHF and standby UHF are all lost:

- 8. IFF/SSR Squawk 7600
- 9. Transmit blind
- 10. Listen out on ILS ident
- 11. Carry out loss of communications procedure

MICROPHONE FAILURE

- 1. Helmet and PEC connections.....Check
- 2. IFF/SSR Squawk 7600
- 3. Carry out speechless procedure (Use TONE button on main UHF radio)

FAILURE TO TRANSMIT

If metallic click on PTT but no carrier wave heard with intercom normal:

1. Use alternative radio

If no metallic click or carrier wave heard:

1. Alt PTT..... Select and use

If fault remains:

2. Carry out MICROPHONE FAILURE (above)

Trims Comms

E-25

LSHC (overleaf)

E-25

LOW SPEED HANDLING CHECK

CAUTION: Do not decrease airspeed to less than threshold speed or a speed at which half aileron or half rudder deflection is required to maintain balanced flight. This is the minimum control speed (MCS)

1.	Radio	Distress call. Consider agency,
		intentions, airborne inspection
2.	Setup	Consider height (ideally 5000ft
		AGL) vs remaining VMC
3.	Position	Consider proximity to airfield
		and habitation

Explore handling in pitch and bank after each of the following steps. Note AOB limit and MCS if applicable.

4.	Speed	Decrease slowly to downwind
		speed. Use height to regain
		speed/control if necessary
5.	Airspeed	Below 200 kts
6.	Landing gear	DOWN, confirm 3 greens
7.	Flaps	MID
8.	Flaps	DOWN

Fly a suitable approach profile to not less than threshold speed, or MCS plus 10 kts, and within the AOB limits tested.

NOTE 1: If handling problems encountered due to deceleration, note MCS and fly approach at MCS plus 10 kts. Do not touchdown at less than MCS

NOTE 2: If handling problems encountered due to flap selection, reverse selection.

NOTE 3: If control cannot be regained - EJECT